

Page No. _____

AIRCRAFT LOG BOOK ENTRY

N714DW

S/N 15079105

DATE

Tach Time: 7309.6

Date: May 03, 2017

Aircraft Total Time: 7309.6

Clean aircraft interior. Cleaned and washed aircraft. Removed corrosion treated and prime and painted. Label secondary VOR Indicator INOP. Lubricated all pulleys and hinges. Performed monthly inspection of fire extinguisher in accordance with Amerex manual 0560, initial weight 4lbs. 10.75oz., current weight 4lbs. 10.75oz.

Complied with AD 80-11-04, Nutplate inspection, no defects noted at this time. Next inspection due within 100 hour time in service.

Complied with AD 2011-10-09, Seat tracks and seats, by inspection. No defects noted. Next inspection due within 100 hours time in service or 12 months, whichever occurs first.

Checked ELT for function I/A/W FAR 91.207(d). Found to be OK. Battery expires 3/18.

This aircraft was inspected in scope and detail as set forth in 14 CFR part 43, Appendix (D).

I certify that this aircraft has been inspected in accordance with an Annual Inspection and was determined to be in airworthy condition at this time.

DATE: May 3, 2017

SIGNATURE

IA

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ENGINE LOG BOOK ENTRY

Continental O-200-AS/N 205663-70A
N714DW

Page

Tach Time: 7309.6
Engine Total Time: 5594.4
TSMO: 2217.2

Date: May 3, 2017

K

Install new BA4108 air filter element. Checked torque on vacuum pump mounting nuts all within limits. Replaced exhaust #1 and #3 exhaust gasket. Replaced #1 exhaust clamp hardware. Replaced screw on left exhaust shroud. Checked accessories case bolts for proper torque. Checked oil pan for proper torque, all within specs. Drained engine oil and removed oil filter. Changed oil. Added 6 quarts Aeroshell 100W AD oil. Cleaned Challenger CP48108-C oil filter. Screen checked for contaminants. None detected. Engine run up and checked for leaks. None detected. Compression checked: #1 76/80; #2 80/80; #3 71/80; #4 77/80. Checked engine case for proper torque. This engine was inspected in scope and detail as set forth in 14 CFR Part 43, Appendix (D).

I certify that this engine has been inspected in accordance with a 100 Hour Inspection and was determined to be in airworthy condition at this time.

DATE: MAY 3 2017SIGNED:  FA

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PROPELLER LOG BOOK ENTRY

McCauley 1A102OCM6948

S/N K17682
N714DW

DATE

Tach Time: 7115.9
Propeller Total Time: 7115.9
TSMO: 2078.1

Date: September 22, 2016

WORK

Inspected blades and hub for erosion, corrosion and cracks. None detected. Leading edge filed and sanded smooth as required.
This propeller was inspected in scope and detail as set forth in 14 CFR Part 43, Appendix (D).
I certify that this propeller has been inspected in accordance with a 100 Hour Inspection and was determined to be in airworthy condition at this time.

DATE: 9-22-16

SIGNED: [REDACTED]

PROPELLER LOG BOOK ENTRY

McCauley 1A102OCM6948

S/N K17682
N714DW

Tach Time: 7210.9
Propeller Total Time: 7210.9
TSMO: 2173.1

Date: November 09, 2016

Inspected blades and hub for erosion, corrosion and cracks. None detected. Leading edge filed and sanded smooth as required.
This propeller was inspected in scope and detail as set forth in 14 CFR Part 43, Appendix (D).
I certify that this propeller has been inspected in accordance with a 100 Hour Inspection and was determined to be in airworthy condition at this time.

DATE: NOV 9, 2016

SIGNED: [REDACTED]

PROPELLER LOG BOOK ENTRY

McCauley 1A102OCM6948

S/N K17682
N714DW

Tach Time: 7309.6
Propeller Total Time: 7309.6
TSMO: 2271.8

Date: May 03, 2017

Inspected blades and hub for erosion, corrosion and cracks. None detected. Leading edge filed and sanded smooth as required.
This propeller was inspected in scope and detail as set forth in 14 CFR Part 43, Appendix (D).
I certify that this propeller has been inspected in accordance with a 100 Hour Inspection and was determined to be in airworthy condition at this time.

DATE: MAY 3, 2017

SIGNED: [REDACTED]

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TTS17

TTS10W - 0 -

DATE	RECORDING TACH TIME	TODAYS FLIGHT	TOTAL TIME IN SERVICE	DESCRIPTION OF INSPECTIONS, TESTS, REPAIRS AND ALTERATIONS ENTRIES MUST BE ENDORSED WITH NAME, RATING AND CERTIFICATE NUMBER OF MECHANIC OR REPAIR FACILITY. (SEE BACK PAGES FOR OTHER SPECIFIC ENTRIES.)
2008				
5/15	6089	1000	4200	Date 5/15/2008 Tach Time 6089.0 205663-70A Replaced Cylinders with new ECI cylinders class 61.10 serial numbers 1- 54349-04, 2- 5439-11, 3- 54349-13, 4- 54349-14. New Superior SA530830 FAA-PMA REV L K060044076 piston pins (4) dated 10/24/06 were used. All new gaskets and seals were included in cylinder kits. All work done IAW STC SE10129SC see completed 337 forms dated May 16, 2008. Operational check satisfactory. No change to weight and balance. [REDACTED] Walter R Dix, Jr [REDACTED]
12/17/2008				R+R CARBURETOR WITH RESULTS ENDORSED PN 10-4894-1 SN BE-23-17666 SEE RELEASE CERT # BE-23-17666 OPCHER SATISFACTORY [REDACTED] ALP EA.

(See continuation sheet 3 of 3)

determined by the installer. Engines modified in accordance with this STC will not differ in performance or operating limitations for unmodified engines.

[illegible]

REMOVED CONTINENTAL O-200-A S/N 205663-70A FOR COMPLETE OVERHAUL. CRANKSHAFT REGROUND, RODS AND MAINS-.010. SEE SERVICEABLE TAG. CAMSHAFT, TAPPET BODIES, ROCKER ARMS, AND CONNECTING RODS, DONE BY AIRCRAFT ENGINE & ACCESSORY, SEE SERVICEABLE TAGS. INSTALLED 4 NEW MILLENNIUM CYLS. SUPPLIED BY SUPERIOR. CRANKCASE REPAIRED & CERTIFIED BY AJAX AVIATION INC. SEE SERVICEABLE TAG. 1 NEW MAIN BRG. SET P/N SA530058-A6M10. 1 NEW CON. ROD. BRG. SET P/N SA639640-M10. 1 NEW OIL PUMP IMPELLER P/N SA22354. 1 NEW OIL PUMP IMPELLER P/N SA22355. 1 NEW OIL PUMP COVER P/N SA23410. 1 NEW SET CON ROD BOLTS, NUTS, & PINS, P/N SA 530213, SA24804, & MS24665-227. 1 GASKET SET P/N SA24890-A5B 1 NEW NOSE CASE SEAL P/N SA530019. 8 NEW CHAMPION REM-40E PLUGS. C/w AD. 93-18-03 By Replacing 2 Pcs. VENTURI w/ 1 PC

ALL WORK ACCOMPLISHED IN ACCORDANCE WITH, TELEDYNE CONT. MOTORS O-200 OVERHAUL MANUAL, CESSNA 150 SERVICE MANUAL, & SUPPLEMENTAL TYPE CERTIFICATE NUMBER SE8675SW DATED 8-24-93. REINSTALLED THIS ENGINE IN AIRCRAFT CESSNA 150 S/N 15079105, N714DW.

TACH. 5067.9 8-9-96 T.T. 3180.9