

Docket No. SA-542

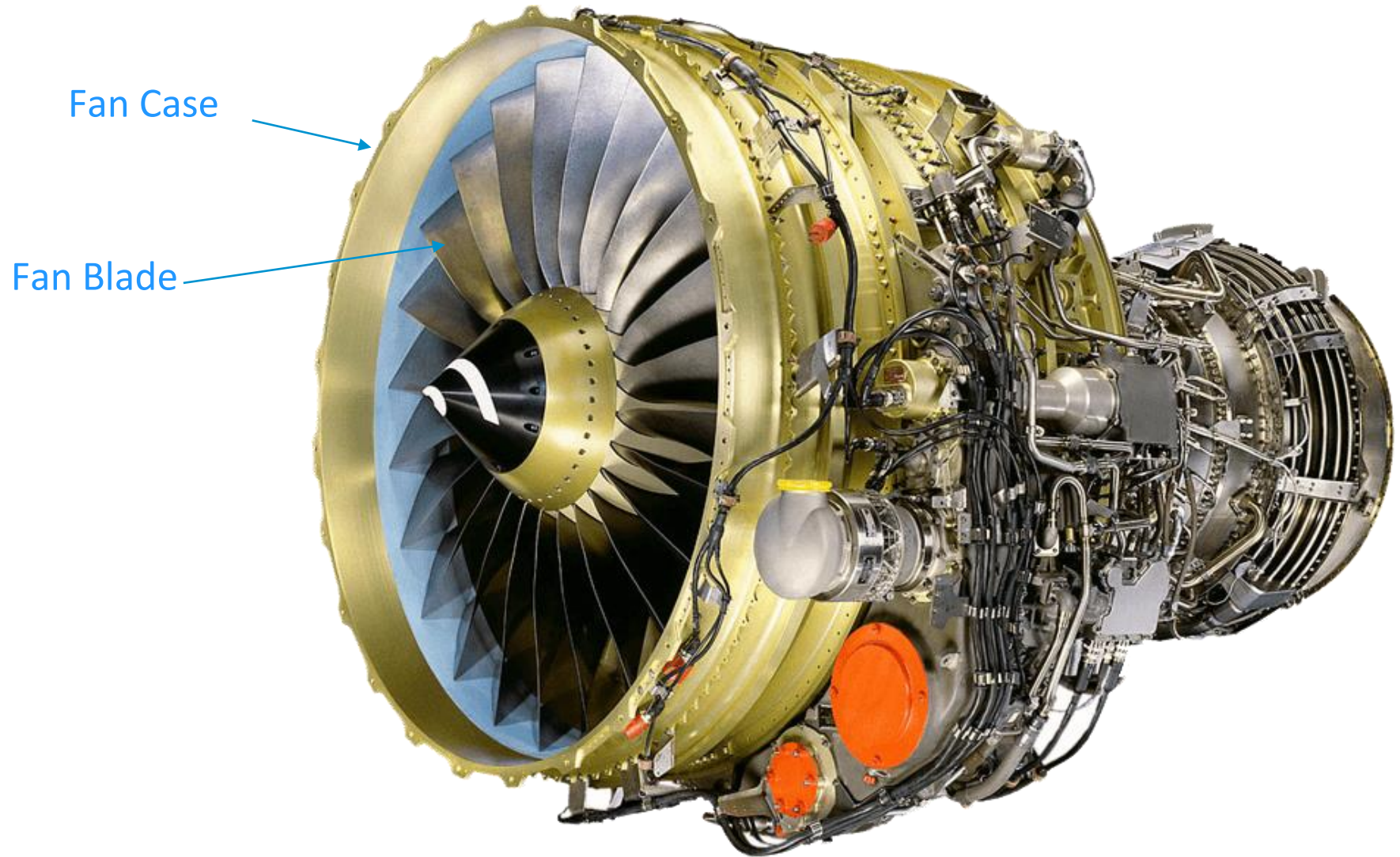
Exhibit No.: 8-B

NATIONAL TRANSPORTATION SAFETY BOARD

Washington, D.C.

CFM Supporting Material
(13 Pages)

CFM56-7B Engine

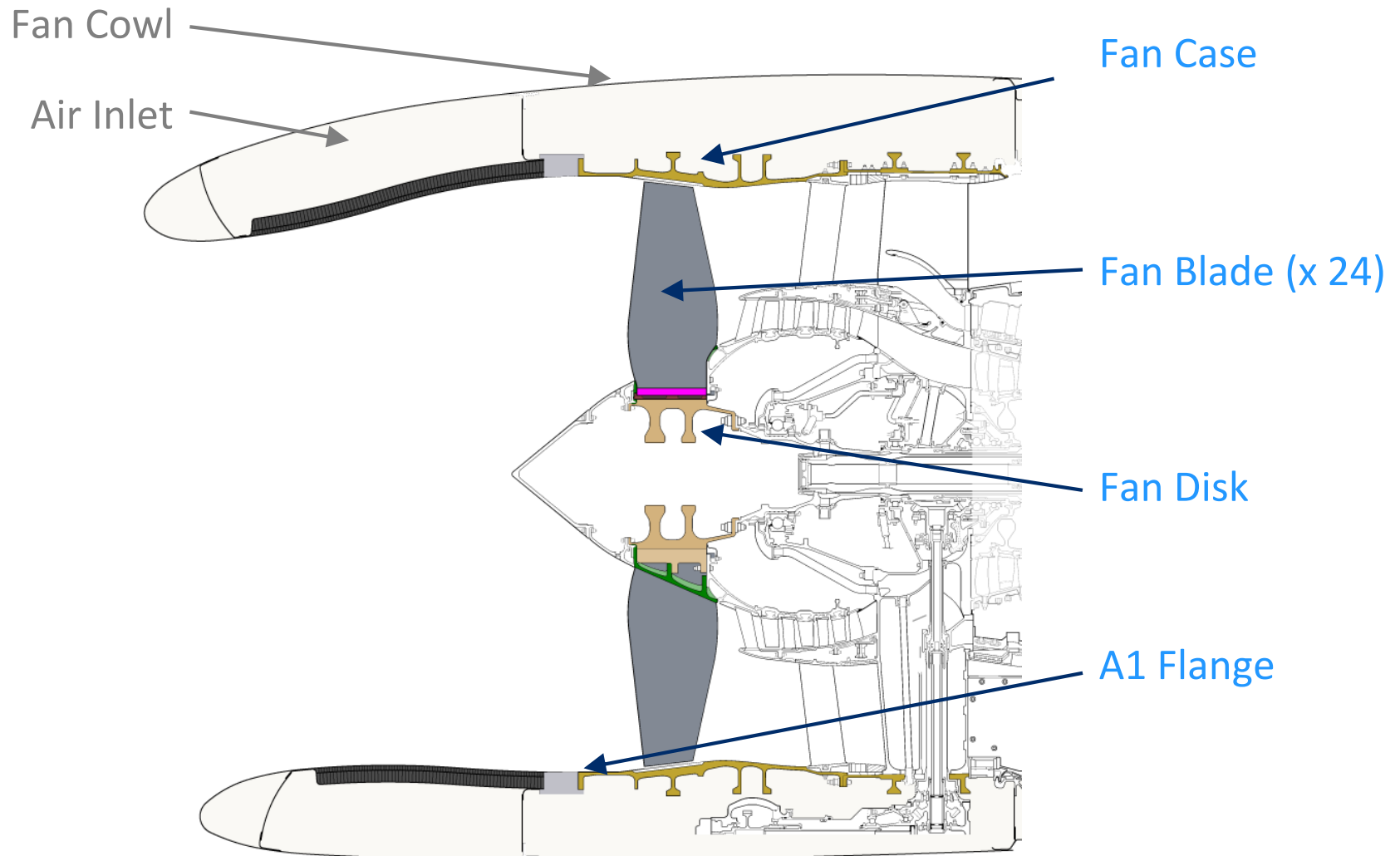


Engine and Nacelle Cross-Section Schematic

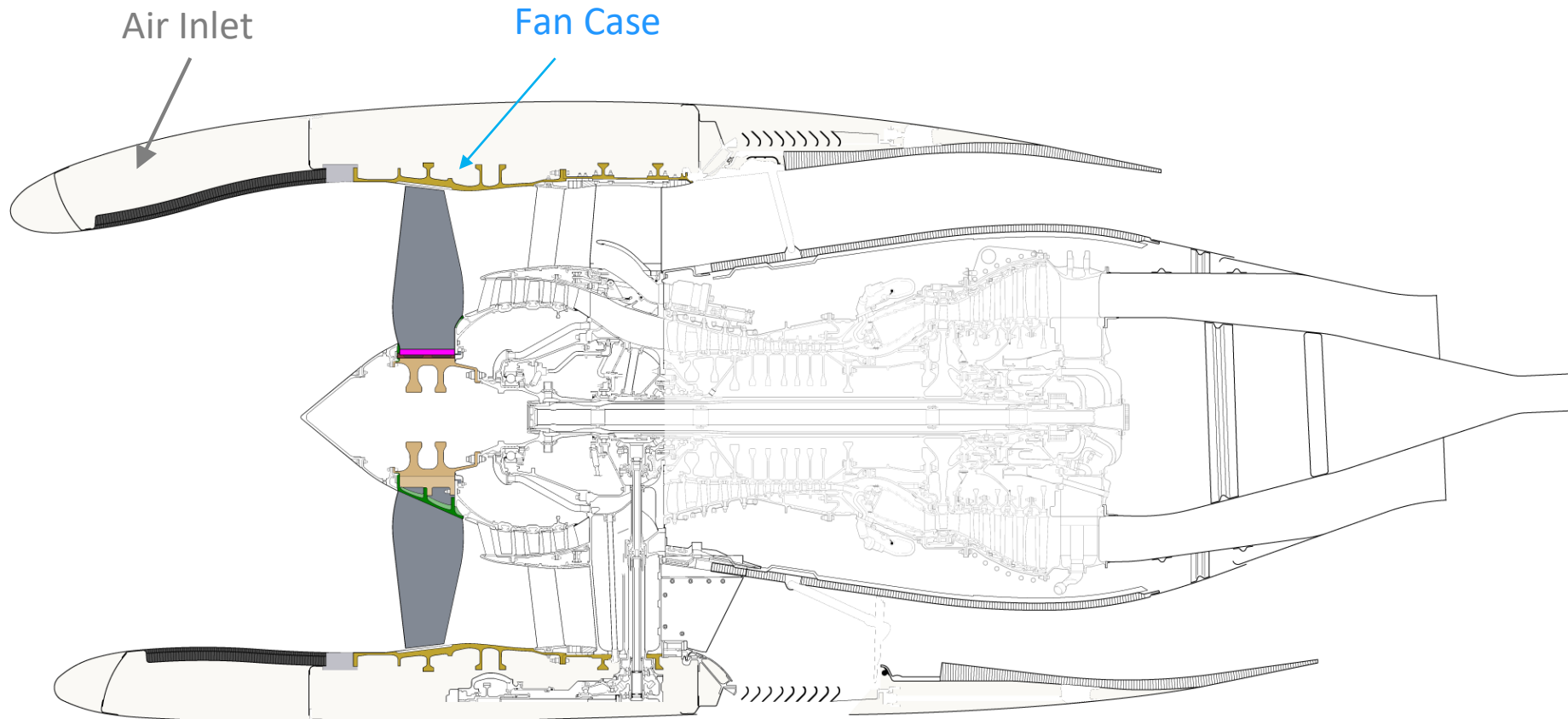


Airframer Hardware

Engine Hardware



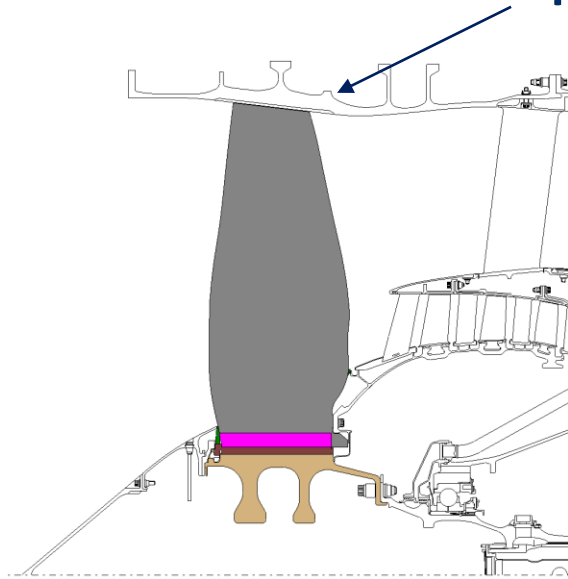
Engine and Nacelle Cross-Section Schematic



Fan Blade / Rotor Assembly



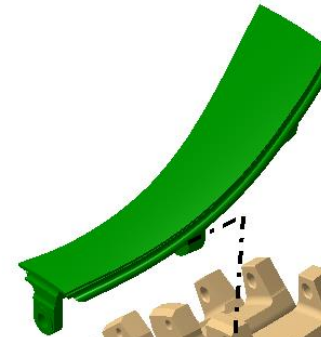
Fan Case



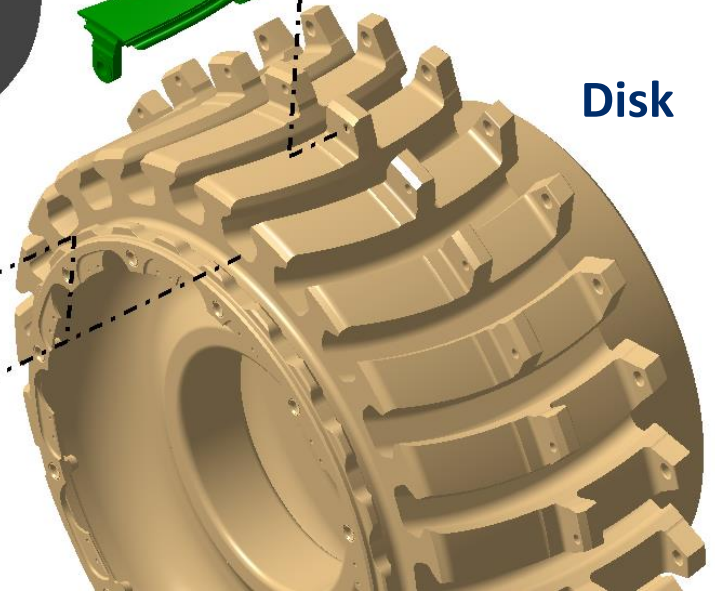
Blade



Platform



Disk



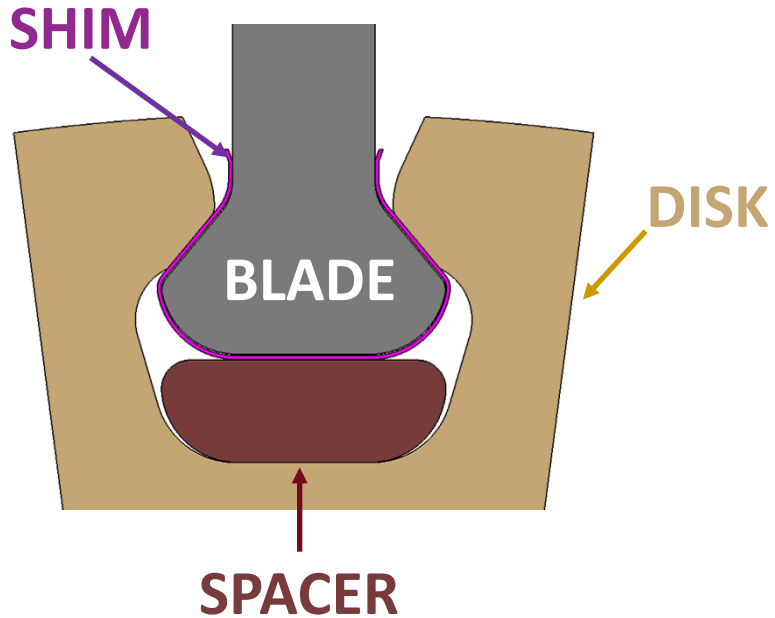
Shim



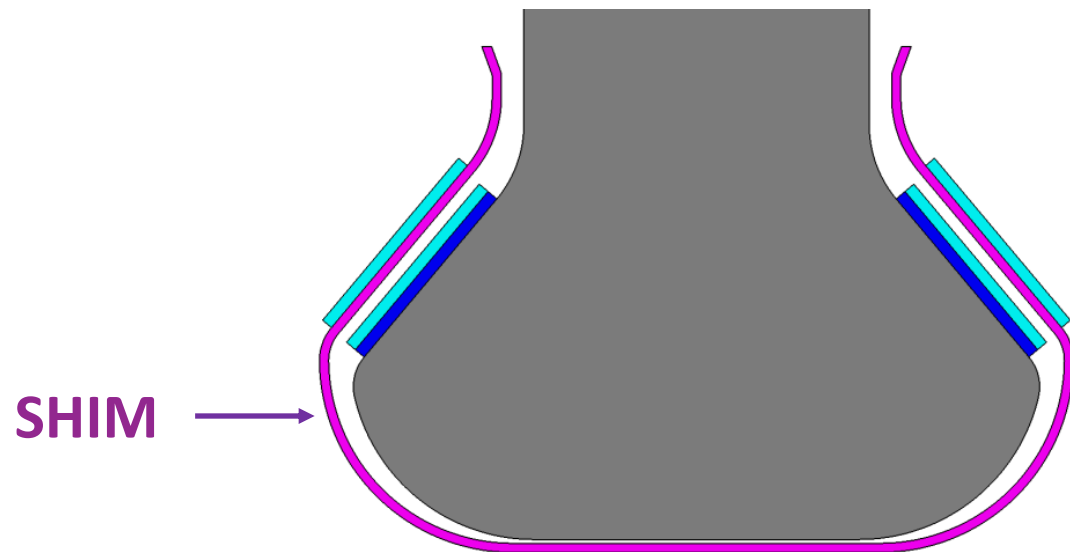
Spacer



Fan Blade Dovetail System



 Lubricant
 Wear coating



Fan Blade Separation



Crack Location

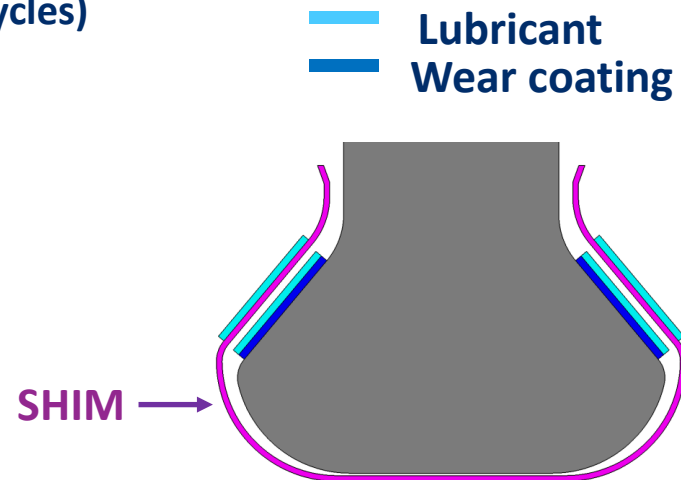
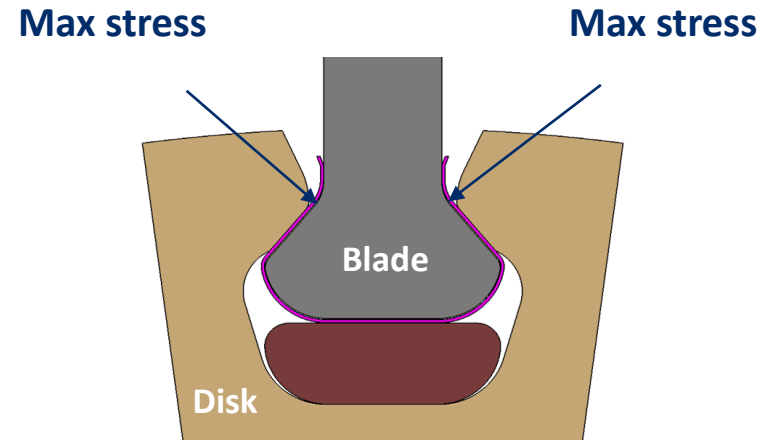
- Maximum stress occurs in root
- Driven by speed & friction

Contributing Factors

- Condition of lubricant & coating
- Material properties
- Length of operating time / quantity of flights (cycles)

Lessons Learned

- Coating durability
- Lubrication interval
- Enhanced inspection at blade dovetail
- Variation in material capability



Fan Blade Comparison



Narrow Chord

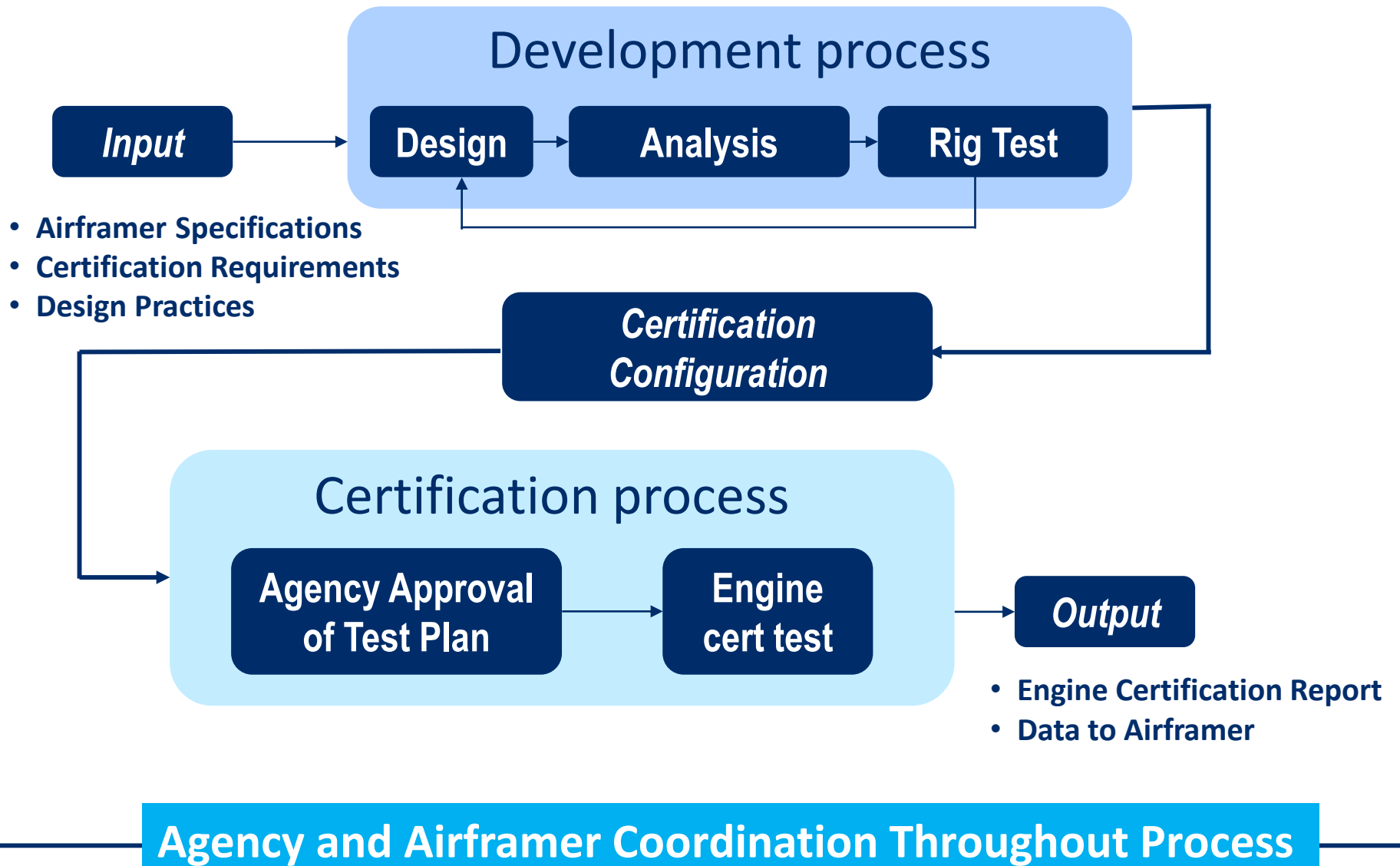


- Improved fuel burn & thrust
- Increased impact capability
- Ease of maintenance

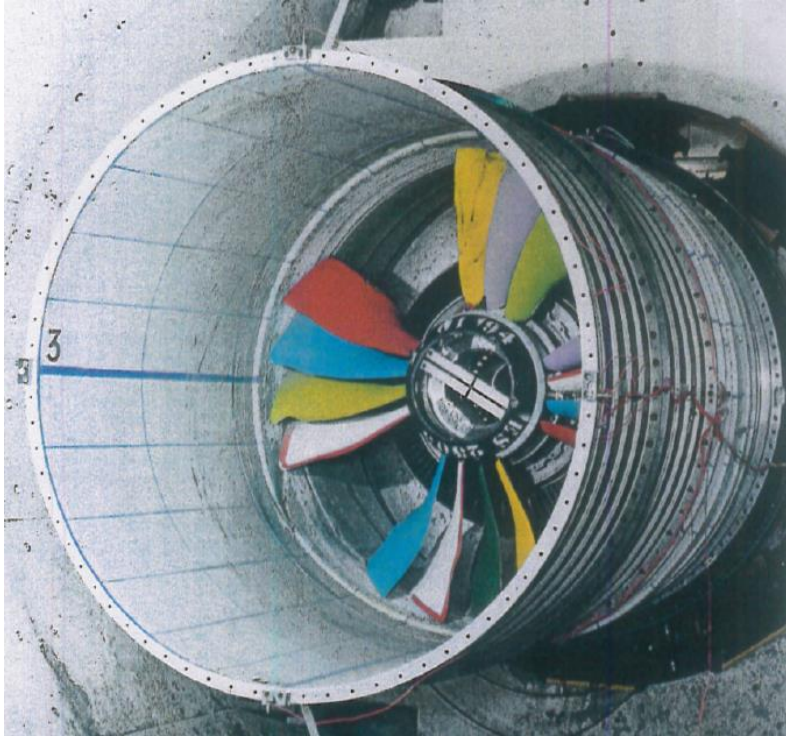
CFM56-7B Wide Chord



Fan Blade Containment Certification Process



Fan Blade Out Rig Test



Test Configuration

- Fan blades
- Fan disk & shaft
- Fan case

Test Conditions

- Electric motor drives the fan
- Run at max speed

Fan Blade Out Certification Engine Test



Test Configuration

- Full engine test
- Certification configuration

Test conditions

- Engine drives fan
- Run at maximum speed

Fan Blade Inspection Tools



Ultrasonic (UTI)



- Inspected in field
- Blade with coating

Eddy Current (ECI)



- Inspected in repair facility
- Blade without coating

Fan Blade After FBO Event



TIP PANEL

**MID-SPAN
FRAGMENTS**

**ROOT
FRAGMENT**



Fan Blade After FBO Event



**Certification
Blade**



**Philadelphia
Blade
(Preliminary)**

