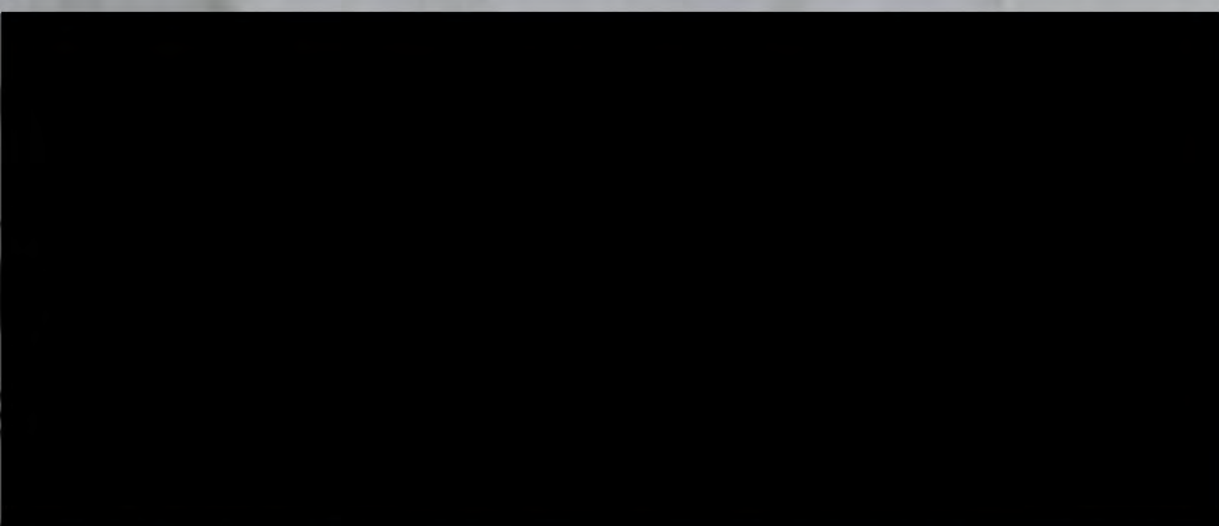
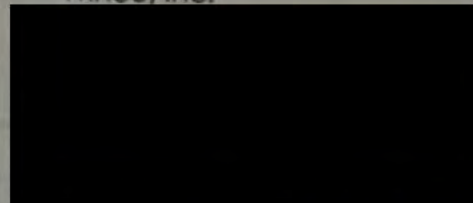




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LES TO LEAD BY

ce them in a situation beyond their capabilities.

ety of yourself and your crew(s).

each job your crew(s) performs.

ned and performed in the proper sequence.

gh tailboard conference is performed.

proper tools, equipment, material and help

erform their job safely, immediately remove

ons.

ANTA SERVICES COMPANY

our point harness and security
l verify the four point is adjusted to the correct length

the proper use of the four-point harness and litter

ct with litter and do not exceed 40kts

loading Stokes Litter in rear of helicopter

rs and seats

ne litter and restrain with litter restraints

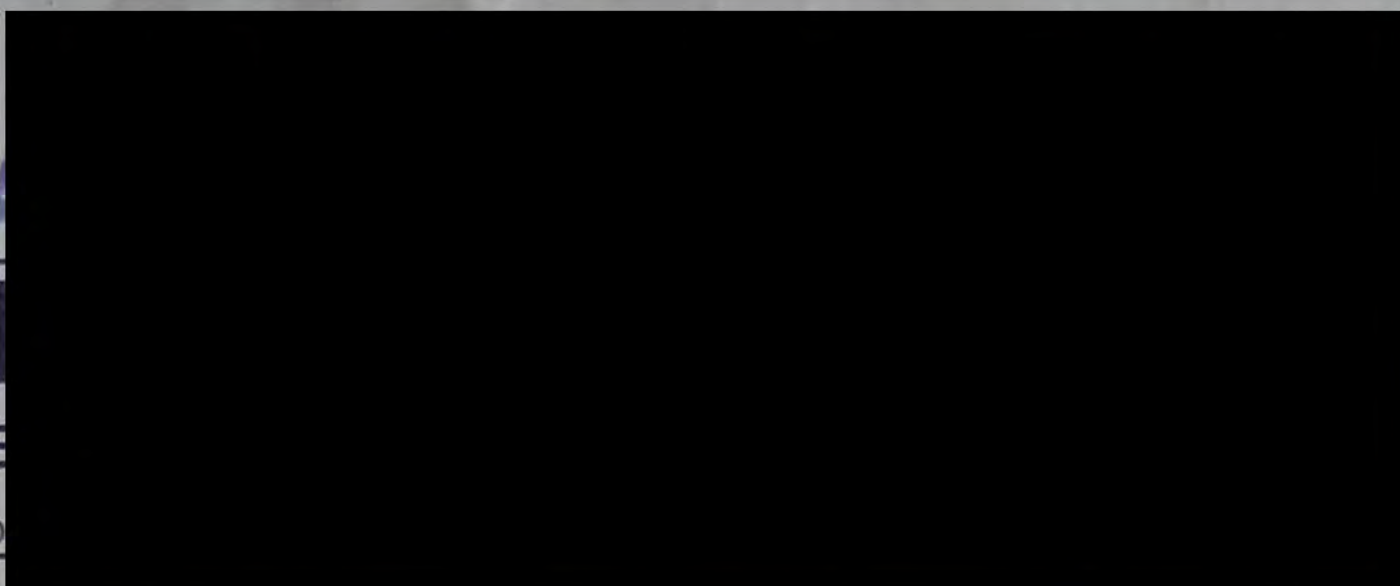
e floor of the helicopter, center it and be sure it is
of the doorframe.

with both rear seat belts

e items

in the litter

HELICOPTER FLIGHT LOG									
AIRCRAFT & ENGINE TIME RECORD					NEXT COMPONENT CHANGE				
AIRCRAFT S/N: 7701590		ENGINE S/N: 295873		COMPONENT		SERIAL NUMBER		DUE @ HOBBS	
Hours To Date: 15132.3		Hours To Date: 3031.3		Start/Gen		6020		15229.3	
Hours This Page:		Hours This Page:		Oil Cooler Brgs.				15229.3	
Total:		Total:							
					PREVIOUS TOTALS		15	15268	12805
DATE	PILOT	DESCRIPTION			HOBBS	TT	CYC	TQE	HOOK
3/13	AB	MJ# 210045			15132.3				
		Pull rope (x 9), fixed ropes HEC			15136.8	4.5	3	24	3.9
3/14	AB	clipped fiber + spacers, skid			15136.8				
					15139.9	3.1	3	15	-
3/15	AB	Pull rope (x1), fix ropes, hung hold downs			15139.9				
					15141.9	2.0	3	12	1.4
3/16	AB	NFW, wind							
3/17	AB								
3/18									
3/19									
WEEKLY TOTALS									
DAILY TOT READINGS					CUMULATIVE TOTALS				
DATE	START	OPERATING	PILOT REMARKS / MECHANICS ACTIONS						
3/13	839/4	669/0	3/16/22 INSTALLED HOBBS AND SWITCH (HOBBS HRS START						
3/14	839/4	707/0	@ 98.2) SEE LOG BOOK ENTRY FOR DETAILS. AP [REDACTED]						
3/15	839/4	707/0	ACT: 15141.9						
3/16			3/16/22 Removed Fuel Control Ducta Starting issues and						
3/17	839/4	707/0	installed serviceable Engine fuel See log book for Details.						
3/18			AP [REDACTED]						
3/19									
TQE INSP DUE: (TQ EVENTS)			N/A	DATE	A.D.'s & SERVICE BULLETINS				DUE @
35 HR TQE INSP DUE: (HOBBS)			N/A						
AIRFRAME 100 HR INSP DUE: (HOBBS)			15224.7	3/13	DN-173, EN-64, FN-52				Pre-Flight
ENGINE 150 HR INSP DUE: (HOBBS)			15268.3		Lead-Lag Link Attach Nuts				
AIRFRAME 300 HR INSP DUE: (HOBBS)			15306.7	3/14	AD88-17-09/R1				Pre-Flight
ENGINE 300 HR INSP DUE: (HOBBS)			15410.7		Inspect T/R Studs				
YEARLY INSP DUE: (M/D/Y)			10/22	3/15	DN-173, EN-64, FN-52				Pre-Flight
					Lead-Lag Link Attach Nuts				
					AD88-17-09/R1				Pre-Flight
					Inspect T/R Studs				
					DN-173, EN-64, FN-52				Pre-Flight
					Lead-Lag Link Attach Nuts				
					AD88-17-09/R1				Pre-Flight
					Inspect T/R Studs				
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					Lead-Lag Link Attach Nuts				
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					Lead-Lag Link Attach Nuts				
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					Inspect T/R Studs				
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					DN-173, EN-64, FN-52				Pre-Flight
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					Inspect T/R Studs				
					DN-173, EN-64, FN-52				



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FIVE RULES TO LEAD BY

Do not place them in a situation beyond their capabilities.

le for the safety of yourself and your crew(s).

active role in each job your crew(s) performs.

work is planned and performed in the proper sequence.

that a thorough tailboard conference is performed.

crew has the proper tools, equipment, material and help b correctly.

s unable to perform their job safely, immediately remove

No Exceptions.

A QUANTA SERVICES COMPANY

ging of four point harness and securing
litter, and verify the four point is adjusted to the correct length
operation.

sonnel on the proper use of the four-point harness and litter

ual contact with litter and do not exceed 40kts

When loading Stokes Litter in rear of helicopter

re rear doors and seats

erson in the litter and restrain with litter restraints

litter on the floor of the helicopter, center it and be sure it is

the back of the doorframe.

the litter with both rear seat belts

any loose items

not further the injuries of the person in the litter

HELICOPTER FLIGHT LOG										
AIRCRAFT & ENGINE TIME RECORD					NEXT COMPONENT CHANGE					
AIRCRAFT S/N: 7701590		ENGINE S/N: 295832		COMPONENT		SERIAL NUMBER		DUE @ HOBBS		
Hours To Date: 15110.7		Hours To Date: 3009.7		start/gen		6020		15229.3		
Hours This Page: 21.6		Hours This Page: 21.6		oil cooler brgs.				15229.3		
Total: 15132.3		Total: 3031.3								
					PREVIOUS TOTALS					
DATE	PILOT	DESCRIPTION			HOBBS	TT	CYC	TQE	HOOK	
3/6	CT	MJ#210045			15110.7					
					15114.4	3.7	4	20	.5	
3/7	AB	Pulled rope(x4), HEC blocks + fix rope, changed fiber block off skid			15114.4					
					15118.7	4.3	3	18	3.5	
3/8	AB	Pulled rope(x3), HEC blocks			15118.7					
					15122.4	3.7	4	26	3.0	
3/9	AB	Hang blocks, clipped fiber			15122.4					
					15124.7	2.3	1	14	0.9	
3/10	AB	Weathered out after 2 attempts			15124.7					
					15126.1	1.4	1	4	-	
3/11	AB	NFW, wind								
3/12	AB/DM	Hang blocks, pulled rope			15126.1					
					15132.3	6.2	2	31	5.0	
					WEEKLY TOTALS		21.6	15	113	12.9
DAILY TOT READINGS					CUMULATIVE TOTALS					
DATE	START	OPERATING	PILOT REMARKS / MECHANICS ACTIONS							
3/6	774/0	669/0	3/9 Act: 15124.7 C/W 100hr Inspection 1AW							
3/7	774/0	669/0	MDHI-CSP-HMI-TL + Winco CMP-1A Rev 4. See							
3/8	824/2	669/0	logbook for details							
3/9	839/4	669/0								
3/10	839/4	669/0								
3/11										
3/12	839/4	669/0								
TQE INSP DUE: (TQ EVENTS)			N/A	DATE	A.D.'s & SERVICE BULLETINS				DUE @	
35 HR TQE INSP DUE: (HOBBS)			N/A							
AIRFRAME 100 HR INSP DUE: (HOBBS)			15224.7	3/6	DN-173, EN-64, FN-52 Lead-Lag Link Attach Nuts				Pre-Flight	
ENGINE 150 HR INSP DUE: (HOBBS)			15260.7	3/6	AD88-17-09/R1 Inspect T/R Studs					
AIRFRAME 300 HR INSP DUE: (HOBBS)			15306.7	3/7	DN-173, EN-64, FN-52 Lead-Lag Link Attach Nuts				Pre-Flight	
ENGINE 300 HR INSP DUE: (HOBBS)			15420.7	3/7	AD88-17-09/R1 Inspect T/R Studs				Pre-Flight	
YEARLY INSP DUE: (M/D/Y)			10/22	3/8	DN-173, EN-64, FN-52 Lead-Lag Link Attach Nuts				Pre-Flight	
				3/8	AD88-17-09/R1 Inspect T/R Studs				Pre-Flight	
				3/8	DN-173, EN-64, FN-52 Lead-Lag Link Attach Nuts				Pre-Flight	
				3/8	AD88-17-09/R1 Inspect T/R Studs				Pre-Flight	
				3/9	DN-173, EN-64, FN-52 Lead-Lag Link Attach Nuts				Pre-Flight	
				3/9	AD88-17-09/R1 Inspect T/R Studs				Pre-Flight	
				3/9	DN-173, EN-64, FN-52 Lead-Lag Link Attach Nuts				Pre-Flight	
				3/9	AD88-17-09/R1 Inspect T/R Studs				Pre-Flight	
				3/10	DN-173, EN-64, FN-52 Lead-Lag Link Attach Nuts				Pre-Flight	
				3/10	AD88-17-09/R1 Inspect T/R Studs				Pre-Flight	
				3/10	DN-173, EN-64, FN-52 Lead-Lag Link Attach Nuts				Pre-Flight	
				3/10	AD88-17-09/R1 Inspect T/R Studs				Pre-Flight	
				3/11	DN-173, EN-64, FN-52 Lead-Lag Link Attach Nuts				Pre-Flight	
				3/11	AD88-17-09/R1 Inspect T/R Studs				Pre-Flight	
				3/11	DN-173, EN-64, FN-52 Lead-Lag Link Attach Nuts				Pre-Flight	
				3/11	AD88-17-09/R1 Inspect T/R Studs				Pre-Flight	
				3/12	DN-173, EN-64, FN-52 Lead-Lag Link Attach Nuts				Pre-Flight	
				3/12	AD88-17-09/R1 Inspect T/R Studs				Pre-Flight	
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				3/12	DN-173, EN-64, FN-52 Lead-Lag Link Attach Nuts				Pre-Flight	
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				3/12	DN-173, EN-64, FN-52 Lead-Lag Link Attach Nuts				Pre-Flight	
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				3/12	AD88-17-09/R1 Inspect T/R Studs				Pre-Flight	
				3/12	DN-173, EN-64, FN-52 Lead-Lag Link Attach Nuts				Pre-Flight	
				3/12	AD88-17-09/R1 Inspect T/R Studs				Pre-Flight	
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				3/12						

TO LEAD BY

them in a situation beyond their capabilities.
of yourself and your crew(s).
ch job your crew(s) performs.
and performed in the **proper sequence**.
tailboard conference is performed.
oper tools, equipment, material and help

orm their job safely, **immediately remove**

r point harness and
verify the four point is adjusted to the correct length

the proper use of the four-point harness and litter

with litter and do not exceed 40kts

ading Stokes Litter in rear of helicopter

s and seats
e litter and restrain with litter restraints
floor of the helicopter, center it and be sure it is
f the doorframe.

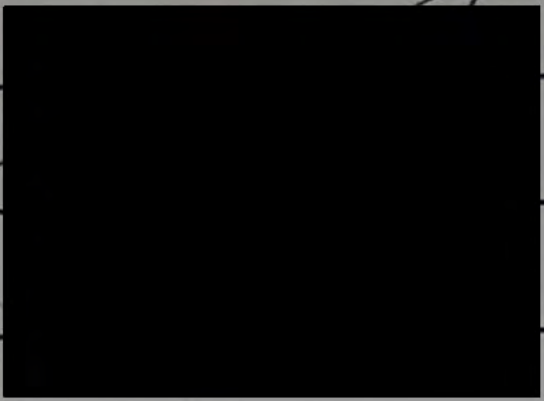
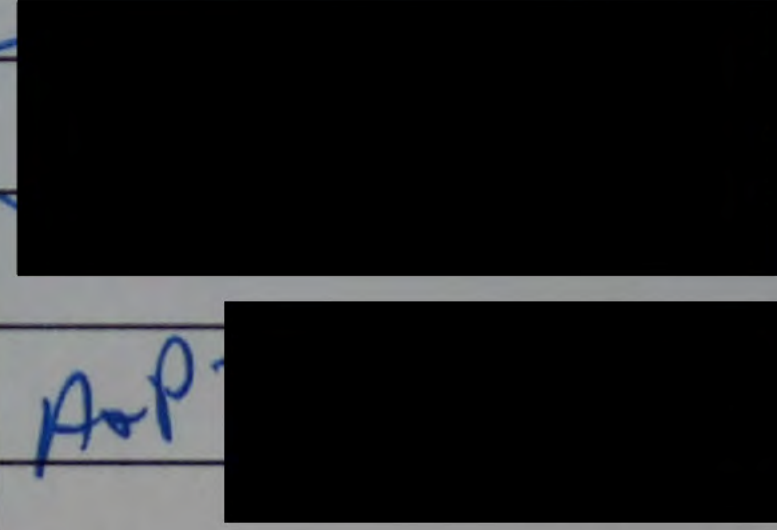
with both rear seat belts

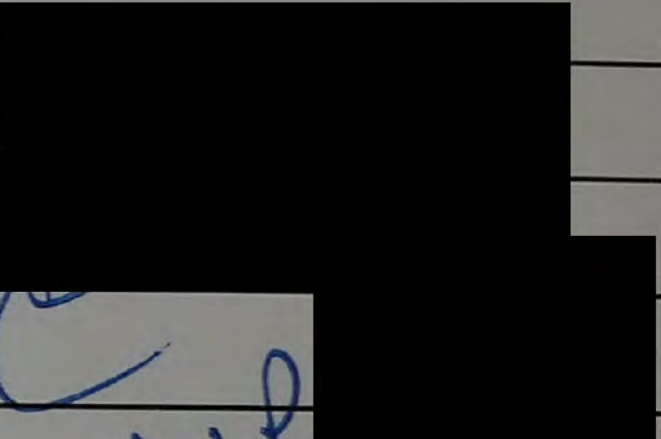
Items
f the person in the litter

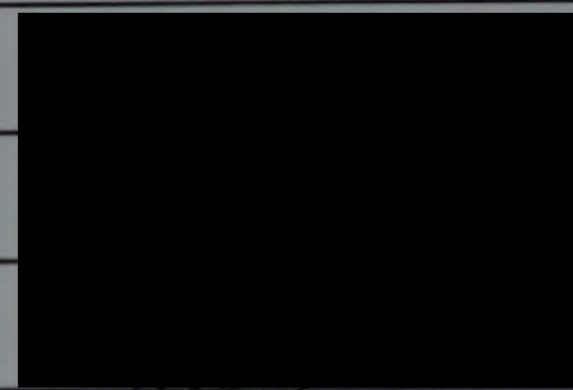
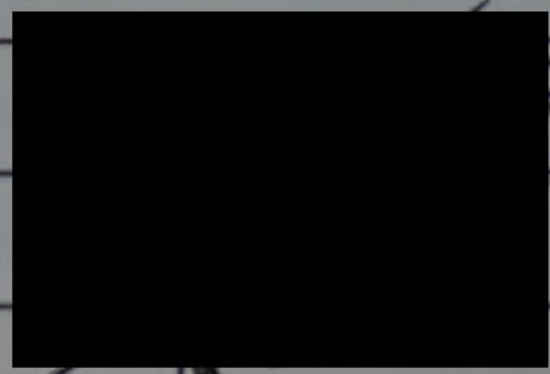
HELICOPTER FLIGHT LOG										
AIRCRAFT & ENGINE TIME RECORD					NEXT COMPONENT CHANGE					
AIRCRAFT S/N: 7701590		ENGINE S/N: 295873		COMPONENT	SERIAL NUMBER	DUE @ HOBBS				
Hours To Date: 15080.3		Hours To Date: 4464.6		Horizontal Stub		15096.6				
Hours This Page: 30.4		Hours This Page: 30.4								
Total: 15110.7		Total: 4495.0								
					PREVIOUS TOTALS		1517	14945	12614	
DATE	PILOT	DESCRIPTION			HOBBS	TT	CYC	TQE	HOOK	
2/27	CT	m5 Job #210045 Carroll Town			15080.3					
2/27	CT	Pull rope 5x10000', Support wire pull HEC, Clip			15085.7	5.4	4	50	2.6	
2/28	CT	De-clip fiber, install blocks, dampers			15085.7					
2/28	CT				15090.2	4.6	4	30		
3/01	CT	Clip fiber, Install Bird Diverters			15090.3					
3/01	CT				15094.9	4.6	3	30		
3/01	CT	Clip fiber, pull ropes for			15094.9					
3/01	CT	conductor + fiber, support pull HEC			15101.7	6.8	4	40	2.5	
3/03	CT	Fiber rope pull, Support rope pull HEC,			15101.7					
3/03	CT	Bird diverters, clip fiber			15106.0	4.3	3	30	0.6	
3/04	CT	Clip fiber, HEC work Remove blocks			15106.0					
3/04	CT				15110.7	4.7	4	30	0.5	
3/05	CT	WFM-Engine Swap			15110.7					
3/05	CT				15110.7	0				
					WEEKLY TOTALS		30.4	22	210	6.2
					CUMULATIVE TOTALS		1531	15155	12676	
DAILY TOT READINGS					PILOT REMARKS / MECHANICS ACTIONS					
DATE	START	OPERATING								
2/27/22	854/1	761/0	3/5 Compressor Due. Removed Engine S/N CAE-295873							
2/28/22	854/1	761/0	and installed Engine S/N CAE-295873 See logbook							
3/1/22	854/1	761/0	See Details [redacted] ADP [redacted]							
3/2/22	854/1	761/0	Next 150 Hr Engine Insp. @ 15260.7 300 Hr Engine Insp.							
3/3/22	854/1	761/0	Due @ 15410.7							
3/4/22	854/1	761/0								
3/5/22	854/1	761/0								
TQE INSP DUE: (TQ EVENTS)			DATE	A.D.'s & SERVICE BULLETINS			DUE @			
35 HR TQE INSP DUE: (HOBBS)										
AIRFRAME 100 HR INSP DUE: (HOBBS)			15128.4	2/27	DN-173, EN-64, FN-52 Lead-Lag Link Attach Nuts					
ENGINE 150 HR INSP DUE: (HOBBS)			15260.7	2/27	AD88-17-09/R1 Inspect T/R Studs					
AIRFRAME 300 HR INSP DUE: (HOBBS)			15300.7	2/28	DN-173, EN-64, FN-52 Lead-Lag Link Attach Nuts					
ENGINE 300 HR INSP DUE: (HOBBS)			15410.7	2/28	AD88-17-09/R1 Inspect T/R Studs					
YEARLY INSP DUE: (M/D/Y)			10/22	3/1	DN-173, EN-64, FN-52 Lead-Lag Link Attach Nuts					
POWER CHECK										
OAT	P.A.	TQ	TOT	+/-	IDLE					
1	2000	70	615	-38	509/17	3/2	DN-173, EN-64, FN-52 Lead-Lag Link Attach Nuts			
17	2000	70	660	-45	552/20	3/3	AD88-17-09/R1 Inspect T/R Studs			
17	2000	70	659	-46	539/13	3/3	DN-173, EN-64, FN-52 Lead-Lag Link Attach Nuts			
17	2000	70	659	-46	508/15	3/4	AD88-17-09/R1 Inspect T/R Studs			
1	2000	70	614	-39	506/14	3/4	DN-173, EN-64, FN-52 Lead-Lag Link Attach Nuts			
11	2000	70	643	-42	528/12	3/5	AD88-17-09/R1 Inspect T/R Studs			
FROM (Hobbs)			TO (Hobbs)		WEEK ENDING DATE	AIRCRAFT REGISTRATION		MODEL		
15080.3			15110.7		3/5/22	N500MZ		369D		

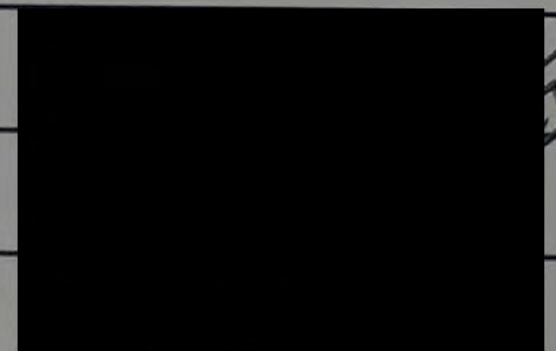
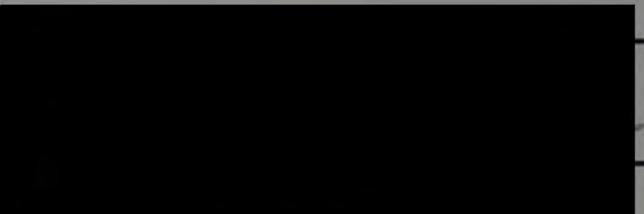
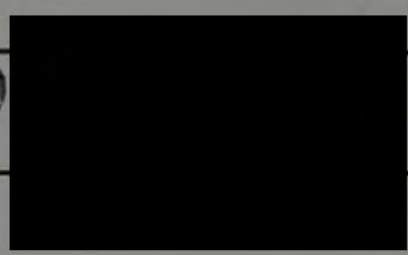
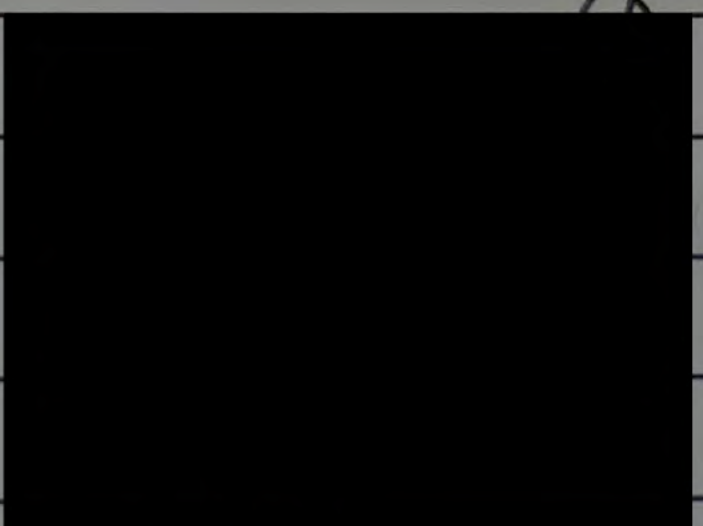
FILL IN DAILY - EMAIL TO OFFICE EVERY SUNDAY - MAIL ENTIRE BOOK TO OFFICE UPON COMPLETION

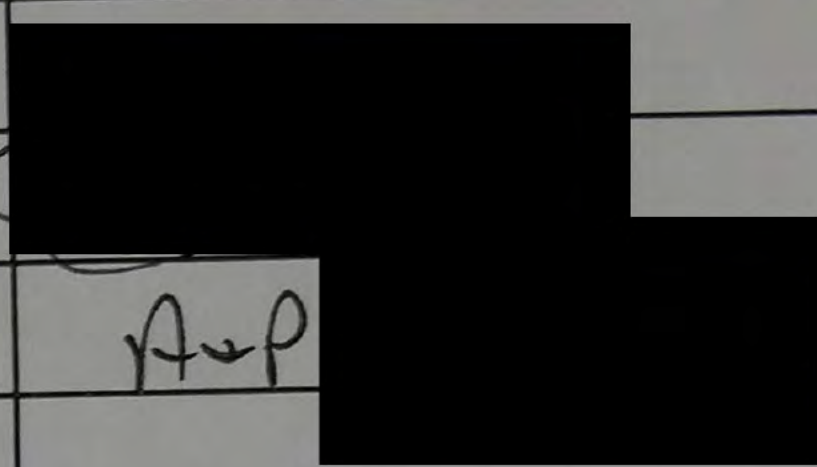
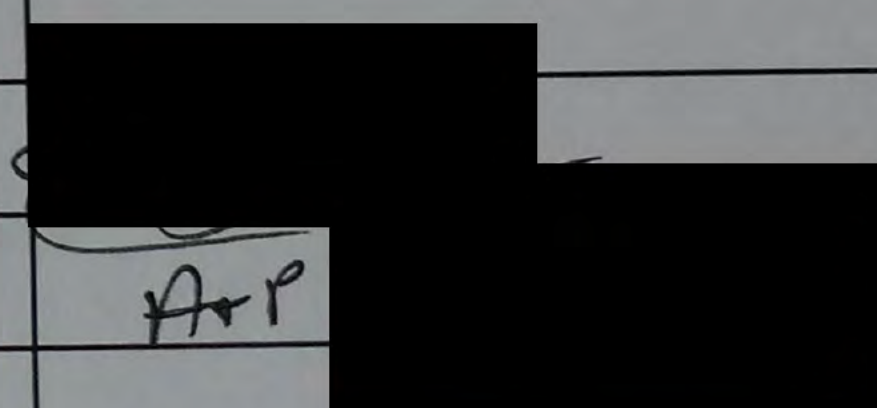
WHITE - OFFICE COPY CANARY - FIELD COPY

MDHC MODEL: 369D N500MZ		WINCO INC. ROTORCRAFT LOG	AIRCRAFT S/N: 770159D ENG. S/N: CAE-295683
DATE / ACTT / HOOK-T	MAINTENANCE PERFORMED		SIGNATURE / LIC. #
3/9/2022 Actt: 15124.7 Cyc: 1550 TQE: 15219	Performed 100 Hr Airframe Inspection IAW MDHI-CSP-HMI-II 100 Hr Inspection Check list and Winco CMP-1A Rev. 4 Complied with Recurring AD'S see logbook for details, Dragn and Reserviced T/R G/B. Lubed splines and Regreased upper bearings on O/R clutch. T/R balance check .02 @ 70% .03 @ 100%. All work performed IAW MDHI-CSP-HMI-II M.M. I certify this Aircraft is in a Airworthy condition at this time		 AP
3/16/22 ACTT: 15141.9	INSTALLED HOBBS METER PN: 100689 (98.2 HOURS @ INSTAL) INSTALLED SWITCH PN: AN 3021-3 IN LINE WITH HOBBS. OP'S CHECK PERFORMED AT THIS TIME.		 AP
3/5/2022 AcTT: 15110.7 EngTT old: 4495.0 EngTT new: 3009.7 dd Cyc: 1539 new cyc: 0	Compressor Dec. Removed Engine Assy P/N 23053265 S/N CAE-295873 and Installed Engine Assy. P/N 23035212 S/N CAE-295832, Governor P/N 2524667-16 S/N 85240068 and Fuel Control P/N 2549865-7 S/N BR53585 was installed on Engine Assy. Perform 150 Hr Inspection IAW Rolls-Royce 250-C20R M.M. Power Check 28 cold. All work perform IAW MDHI-CSP-HMI-II and Rolls-Royce 250-C20R M.M. Performed Runup and leak check. Adjusted idle and Governor Beep Range <u>No Defects Note</u>		 AP

MDHC MODEL: 369D		WINCO INC. ROTORCRAFT LOG	AIRCRAFT S/N: 770159D
N500MZ			ENG. S/N: CAE-295683
DATE / ACTT / HOOK-T	MAINTENANCE PERFORMED		SIGNATURE / LIC. #
1/26/2022 ACTT: 15012.0 CYC: 1480 TQE: 14373	I certify that the 600 HR Main Rotor Drive shaft inspection has been completed & found airworthy IAW MD Helicopters Maintenance Manual, CSP-HMI-2, Section 05-20-20, Table 1. No discrepancies noted.		
2/2/2022 Actt: 15028.4 Cyc: 1486 TQE: 14446	Performed 100 Hr Airframe Inspection IAW MDHI-CSP-HMI-II 100 Hour Inspection Check list and Winco CMP-1A Rev. 4. Complied with Recurring AD's see logbook for details. Drained and Reserviced T/R G/B. Lubed Splines and Regreased upper bearing on ~~10~~ ¹⁰ O/R Clutch. T/R balance check .07 @ 70% and .03 @ 100%. Main Rotor Drive Shaft Splines worn. Removed M/R D/S S/N 5009-0789 and installed S/N 5009-1609. All work performed IAW MDHI-CSP-HMI-II M.M. I certify this Aircraft is in a <u>Airworthy condition at this time</u>		 AJP
3/1/2022 ACTT: 15094.9 CYC: 1528 TQE: 15055	I certify that the high timed horizontal stabilizer was removed PN: 369D23601-511, SN: 1858 & replaced with a new horizontal stabilizer PN: 369X23601-511, SN: 2285 IAW MD Helicopters Maintenance Manual CSP-HMI-2, Section 53-50-10, Para. 3A-3B. Balanced tail rotor to .02 @ 70% & .03 at 100%. No discrepancies noted.		

MDHC MODEL: 369D N500MZ		WINCO INC. ROTORCRAFT LOG		AIRCRAFT S/N: 770159D ENG. S/N: CAE-295683	
DATE / ACTT / HOOK-T		MAINTENANCE PERFORMED		SIGNATURE / LIC. #	
-Cont' from P. 115		Main rotor folding pins installed IAW CSP-HMI-2, section 62-10-00, p. 402. No discrepancies noted. Removed overrunning clutch assembly PN: 369A5350-51, SN: 2222-7698 for 300HR sprag inspection. Installed an airworthy overrunning clutch assembly with completed 300HR sprag inspection PN: 369A5350-41, SN: 4908 IAW AD 90-19-02 & CSP-HMI-2, section 63-10-00, p. 410. AD next due at hook time 1528.4.			
1/22/2022 ACTT: 15006.7 CYC: 1477 TQE: 14313		I certify that the M/R transmission 600HR oil & filter change has been completed & found airworthy IAW MD Helicopters maintenance manual, CSP-HMI-2, section 05-20-20, Table 1. Leak check OK.			
1/26/2022 ACTT: 15012.0 CYC: 1480 TQE: 14373		I certify that the 600HR cyclic system free play test has been completed & found airworthy IAW MD Helicopters Maintenance Manual, CSP-HMI-2, section 05-20-20, Table 1. No discrepancies noted.			

MDHC MODEL: 369D	WINCO INC. ROTORCRAFT LOG	AIRCRAFT S/N: 770159D
N500MZ		ENG. S/N: CAE-295683
DATE / ACTT / HOOK-T	MAINTENANCE PERFORMED	SIGNATURE / LIC. #
1/3/2022	Removed AFS Filters P/N: 102450-101, S/N: 0974 ÷	
ACTT: 14961.16	P/N: 102400-101, S/N: 0983 for cleaning. Installed	
CYC: 1459	freshly cleaned AFS Filters P/N: 102450-101, S/N: 0278 ÷	
TQE: 13981	P/N: 102400-101, S/N: 0215 IAW AFS ICA. No	
	further discrepancies noted.	
1/14/2022	Upper Scissor link Bearings worn out. Removed M/R Upper	
Actt: 14993.0	Scissor-link Bearings and installed New P/N 369A7951-7	
CYC: 1472	op's check good. All work performed IAW MDHI-CSPHMI-II A-P	
TQE 14183	No further discrepancies noted	
1/21/2022	I certify that the 300HR Airframe Inspection has	
ACTT: 15006.7	been completed ÷ found airworthy IAW MD Helicopters	
CYC: 1476	Maintenance Manual CSP-HM-Z, 05-20-10, Table 1 ÷	
TQE: 14313	Winco Inc CMP-3A, Rev. 3. T/R damper was rotated	
	180 degrees during inspection as required IAW CMP-3A.	
	Removed Main rotor folding pins P/N: 369A1004-5, S/N	
	S/Ns: 5510-E491, 5510-E535, 5510-E553, 5510-E974, 5510-E981,	
	5510-E982, 5510-E992, 5510-E998, 5510-F003, ÷ 5510-F009 due to	
	high time. Installed main rotor folding pins P/N: 369A1004-5,	
	S/Ns: 181K523, 181K527, 181K533, 181K538, 181K568, 181K570,	
	181K578, 181K580, 181K588, ÷ 181K689 (TSN: 5585558.0).	
		-Cont' on P. 116

MDHC MODEL: 369D N500MZ		WINCO INC. ROTORCRAFT LOG		AIRCRAFT S/N: 770159D
DATE / ACTT / HOOK-T		MAINTENANCE PERFORMED		SIGNATURE / LIC. #
12/9/2021	Performed 100Hr Airframe Inspection IAW MDHI-CSP-HMI-2	 ARP		
ACTT: 14944.4	and Winco CMP-1A Rev 4. Complied with Recurring			
CYC: 1446	AD's See log Book for Details. Drained and Reserviced			
TQE: 13897	T/R G/B. Lubed Splines and Regreased upper Bearing on			
	O/R Clutch. T/R Balance check .07@ 70% and			
	.02 @ 100%. M/R Scissor link lower Bearing worn.			
	Removed Scissor link Assy. and Installed Serviceable			
	Scissor link Assy. op's check good. I certify this Aircraft			
	is in an Airworthy condition at this time.			
12/9/2021	Complied with SB369D-22R2 by visual Inspection	 ARP		
	and Boroscope Inspection. No Defects Noted. Next Due			
	@ 15244.4			
12/20/2021	Removed aircraft battery PN: R6500, SN: 40897899	 IA		
ACT: 14956.0	due to loss of charge during start up & installed			
CYC: 1457	airworthy aircraft battery, same PN, SN: 41104076			
TQE: 13951	IAW CSP-HMI-3, section 96-05-00. Ops check OK.			

WINCO, INC

Aurora, OR 97002

MD HELICOPTERS MODEL 369D ACTUAL WEIGHT AND BALANCE RECORD

REG. NO.: N500MZ S/N: 770159D DATE WEIGHED: 9/17/2018
 WEIGHED BY: SHANE LAIR A&P
 SCALES USED: M100-3-5CS S/N: 10-6691 CAL. EXPIRES: 3/26/2019

WEIGHING POINTS	SCALE READING (LBS)	TARE OR CALIBRATION CORRECTION (LBS)	NET WEIGHT (LBS)	LONGITUDINAL ARM (INCHES)	LATERAL ARM (INCHES)	LONGITUDINAL MOMENT (INCH-LBS)	LATERAL MOMENT (INCH-LBS)
LEFT MAIN	671.0	0.0	671.0	96.9	-26.0	65019.9	-17446.0
RIGHT MAIN	713.0	0.0	713.0	96.9	26.0	69089.7	18538.0
TAIL	214.0	0.0	214.0	197.2	0.0	42200.8	0.0
TOTAL UNADJUSTED NET WEIGHT			1598.0	110.3	0.7	176310.4	1092.0
TOTAL WEIGHT OF SURPLUS EQUIPMENT			-2.7	95.1	0.0	-256.8	0.0
TOTAL WEIGHT OF MISSING EQUIPMENT			5.9	96.9	-26.7	571.7	-157.5
TOTAL ADJUSTED BASIC WEIGHT			1601.2	110.3	0.6	176625.3	934.5

FUEL/OIL ABOARD AT TIME OF WEIGHING:

	EMPTY	FULL
FUEL	XXXXX	
ENGINE OIL		XXXXX
MAIN GEARBOX		XXXXX
TAIL GEARBOX		XXXXX

MISSING EQUIPMENT AT TIME OF WEIGHING

ITEM	WEIGHT	LONGITUDINAL ARM (INCHES)	LATERAL ARM (INCHES)	LONGITUDINAL MOMENT (INCH-LBS)	LATERAL MOMENT (INCH-LBS)
Safety Handle	4	96.9	-27	387.6	-108
Side pull assy.	1.9	96.9	-26	184.11	-49.4
TOTAL	5.9	96.9	-26.677966	571.71	-157.4

SURPLUS EQUIPMENT AT TIME OF WEIGHING

ITEM	WEIGHT	LONGITUDINAL ARM (INCHES)	LATERAL ARM (INCHES)	LONGITUDINAL MOMENT (INCH-LBS)	LATERAL MOMENT (INCH-LBS)
JACK FITTINGS (2)	1.3	96.9	0.0	126.0	0.0
UNDRAINABLE FUEL	1.4	93.5	0.0	130.9	
TOTAL	2.7	95.1	0.0	256.9	0.0

Model 369D Serial No. 770159D Registration No. N500MZ Date 9/17/2018

EXAMPLE OF FORWARD, AFT, 1700# LOAD AND LATERAL LOADING

EXAMPLE 1, FORWARD

	WEIGHT (LBS)	LONG. ARM (IN.)	LONG. MOMENT (IN. LBS)
BASIC WEIGHT	1601.2	110.31	176625.34
PILOT	170.0	73.50	12495.00
PASSENGER - FWD	170.0	73.50	12495.00
PASSENGER - FWD CENTER	170.0	71.50	12155.00
CRITICAL FUEL QUANTITY	40.0	90.60	3624.00
			0.00
GROSS WEIGHT (CRITICAL FUEL) - FWD C.G.	2151.2	101.06	217394.34

APPROVED FWD C.G. LIMIT FOR EXAMPLE 1 GROSS WEIGHT >99.00 INCHES

EXAMPLE 2, AFT

	WEIGHT (LBS)	LONG. ARM (IN.)	LONG. MOMENT (IN. LBS)
BASIC WEIGHT	1601.2	110.31	176625.34
PILOT	170.0	73.50	12495.00
PASSENGER - AFT L/H	170.0	105.00	17850.00
PASSENGER - AFT R/H	170.0	105.00	17850.00
BAGGAGE - UNDER SEAT	50.0	110.00	5500.00
GROSS WEIGHT (ZERO FUEL) - AFT C.G.	2161.2	106.57	230320.34

APPROVED AFT C.G. LIMIT FOR EXAMPLE 2 GROSS WEIGHT <105.80 INCHES

EXAMPLE 3, 1700# LOAD

	WEIGHT (LBS)	LONG. ARM (IN.)	LONG. MOMENT (IN. LBS)
BASIC WEIGHT	1601.2	110.31	176625.34
PILOT	170.0	73.50	12495.00
LOAD ON HOOK	1700.0	99.00	168300.00
FUEL REQUIRED FOR 5 MIN PICK PLUS 20 MIN RESERVE	80.0	92.75	7420.00
GROSS WEIGHT (REQUIRED FUEL) - AFT C.G.	3551.2	102.74	364840.34

APPROVED AFT C.G. LIMIT FOR EXAMPLE 3 GROSS WEIGHT <103.00 INCHES

EXAMPLE 4, LATERAL

	WEIGHT (LBS)	LAT. ARM (IN.)	LAT. MOMENT (IN. LBS)
BASIC WEIGHT	1601.2	0.6	934.47
PILOT	170.0	-13.0	-2210.0
GROSS WEIGHT (ZERO FUEL) - AFT C.G.	1771.2	-0.7	-1275.53

APPROVED LATERAL C.G LIMIT FOR EXAMPLE 4 GROSS WEIGHT <+3.0 INCHES