

BOSS Aircraft Maintenance

Salisbury NC 28147

United States

FAA Airworthiness Directives Compliance Record

Company:

Category: Airframe

Manufacturer: Cessna Aircraft Company

Model: 182P

Position:

P/N:

S/N: 18262310

Aircraft Registration No: N58807

ATP Revision: 01/26/2022

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
2020-18-01 11/12/2020	[Recurring] To Detect and Address Cracking of the Wing Strut Attach Point. The Unsafe Condition, if Not Addressed, Could Result in Failure of the Wing in Operation, Which Could Result in Loss of Control of the Airplane	01/31/2022 Hrs: 4014 C: 5771.20	C/W par (h) visual Inspection.	Yes	D: 01/30/2025 Hrs: 5014 C: 6771.20	AP [REDACTED] IA/ [REDACTED]
2011-10-09 06/17/2011	[Recurring] To prevent seat slippage or the seat roller housing from departing the seat rail, which may consequently cause, contd.	01/31/2022 Hrs: 4014 C: 5771.20	C/W par (g) inspection.	Yes	D: 01/31/2023 Hrs: 4114 C: 5871.20	AP [REDACTED] IA/ [REDACTED]

Category: Ignition Systems

Manufacturer: ACS Products Company

Model: IGNITION SWITCHES

Position:

P/N:

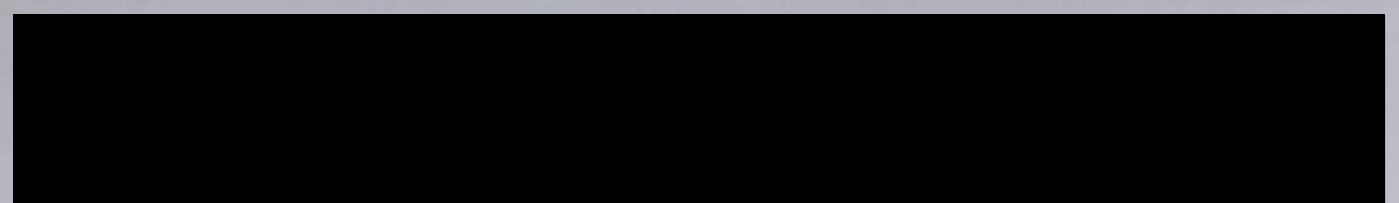
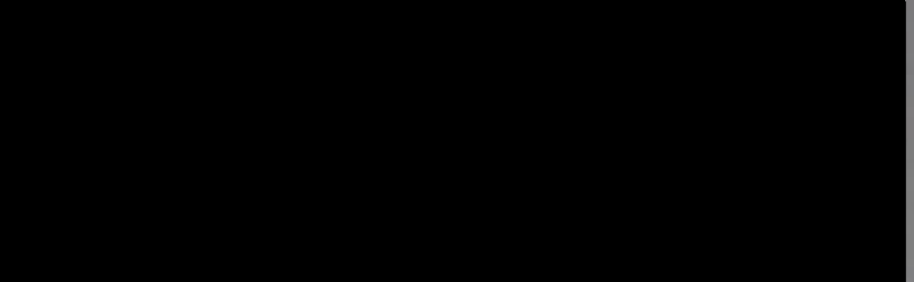
S/N:

ATP Revision: 01/26/2022

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
93-05-06 04/29/1993	[Recurring] TO PREVENT FAILURE OF IGNITION SWITCHES	01/31/2022 Hrs: 4014 C: 5771.20	Installed new Ignition Switch @ Tach: 4041.0/5771.2	Yes	D: -- Hrs: 6014 C: 7771.20	AP [REDACTED] IA/ [REDACTED]



Belle Aircraft Maintenance

Phone 
Fletcher NC 28732

USA

FAA Airworthiness Directives Compliance Record

Company: Mountian Flyers

Aircraft Registration No: N58807

Category: Airframe

Position:

ATP Revision: 12/12/2018

Manufacturer: Cessna Aircraft Company

P/N:

Model: 182P

S/N: 18262310

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
71-21-01 01/01/1971	Superseded by 72-07-09	-- Hrs: -- C: --	-----	No	D: -- Hrs: -- C: --	/
71-22-02 R(1) 11/09/1971	TO DECREASE THE POSSIBILITY OF FAILURE OF THE NOSE GEAR STRUCTURE	-- Hrs: -- C: --	N/A NOT INSTALLED	No	D: -- Hrs: -- C: --	/
72-07-09 10/17/1974	TO DETECT CRACKS AND BOLT LOOSENESS WHICH COULD LEAD TO INFLIGHT SEPARATION OF THE FIN AND THE RUDDER	-- Hrs: -- C: --	N/A BY S/N	No	D: -- Hrs: -- C: --	/
73-17-01 08/16/1973	TO ADVISE THE PILOT CONCERNING PROPER FUEL TRANSFER PUMP OPERATION	-- Hrs: -- C: --	N/A NOT INSTALLED	No	D: -- Hrs: -- C: --	/
73-23-07 11/19/1973	TO PREVENT DEFECTIVE SPAR ATTACHMENT FITTING FROM REMAINING IN SERVICE	-- Hrs: -- C: --	N/A BY S/N	No	D: -- Hrs: -- C: --	/
75-05-02 02/21/1975	TO PREVENT LOSS OF ENGINE OIL, IMPROPER ENGINE LUBRICATION OR ENGINE OIL CONTAMINATION	-- Hrs: -- C: --	N/A NOT INSTALLED	No	D: -- Hrs: -- C: --	/

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
75-16-01 07/30/1975	TO PRECLUDE INADVERTENT FUEL EXHAUSTION DUE TO INCORRECT FUEL PLACARDED CAPACITIES	08/12/1980 Hrs: -- C: --	PCW	No	D: -- Hrs: RT 7/1/93 C: --	/
76-04-03 02/26/1976	TO PRECLUDE RESTRICTIONS OF CONTROL MOVEMENT DUE TO JAMMING OF THE ARC PA-500A ACTUATOR GEAR TRAIN	-- Hrs: -- C: --	N/A NOT INSTALLED	No	D: -- Hrs: -- C: --	/
77-02-09 02/03/1977	WING FLAP ACTUATOR DATE CODE STAMP	-- Hrs: -- C: --	N/A BY S/N	No	D: -- Hrs: -- C: --	/
77-04-05 03/01/1977	TO PREVENT INGESTION OF THE INDUCTION AIR BOX SEAL INTO THE CARBURETOR	07/01/1993 Hrs: 914.4 C: --	PCW	No	D: -- Hrs: -- C: --	/
77-12-08 06/27/1977	TO PREVENT UNWANTED PROPELLER ROTATION	-- Hrs: -- C: --	N/A BY S/N	No	D: -- Hrs: -- C: --	/
77-14-09 07/18/1977	TO ASSURE AIRCRAFT CONTROLLABILITY WHEN OPERATING AT MAXIMUM GROSS WEIGHT AND AFT CENTER OF GRAVITY LOCATION	-- Hrs: -- C: --	N/A BY S/N	No	D: -- Hrs: -- C: --	/
77-23-11 11/28/1977	TO PRECLUDE THE POSSIBILITY OF AN IN-FLIGHT FIRE DUE TO A LOOSE ELT ANTENNA COAXIAL CABLE CONNECTOR, CONTD.	-- Hrs: -- C: --	N/A NOT INSTALLED	No	D: -- Hrs: -- C: --	/
78-01-14 01/23/1978	TO PRECLUDE THE POSSIBILITY OF AN IN-FLIGHT FIRE DUE TO A LOOSE ELT ANTENNA COAXIAL CABLE CONNECTOR, CONTD.	04/12/1979 Hrs: 1192.2 C: --	PCW	No	D: -- Hrs: -- C: --	/
78-26-09 01/01/1978	Superseded by 79-10-14	-- Hrs: -- C: --	-----	No	D: -- Hrs: -- C: --	/

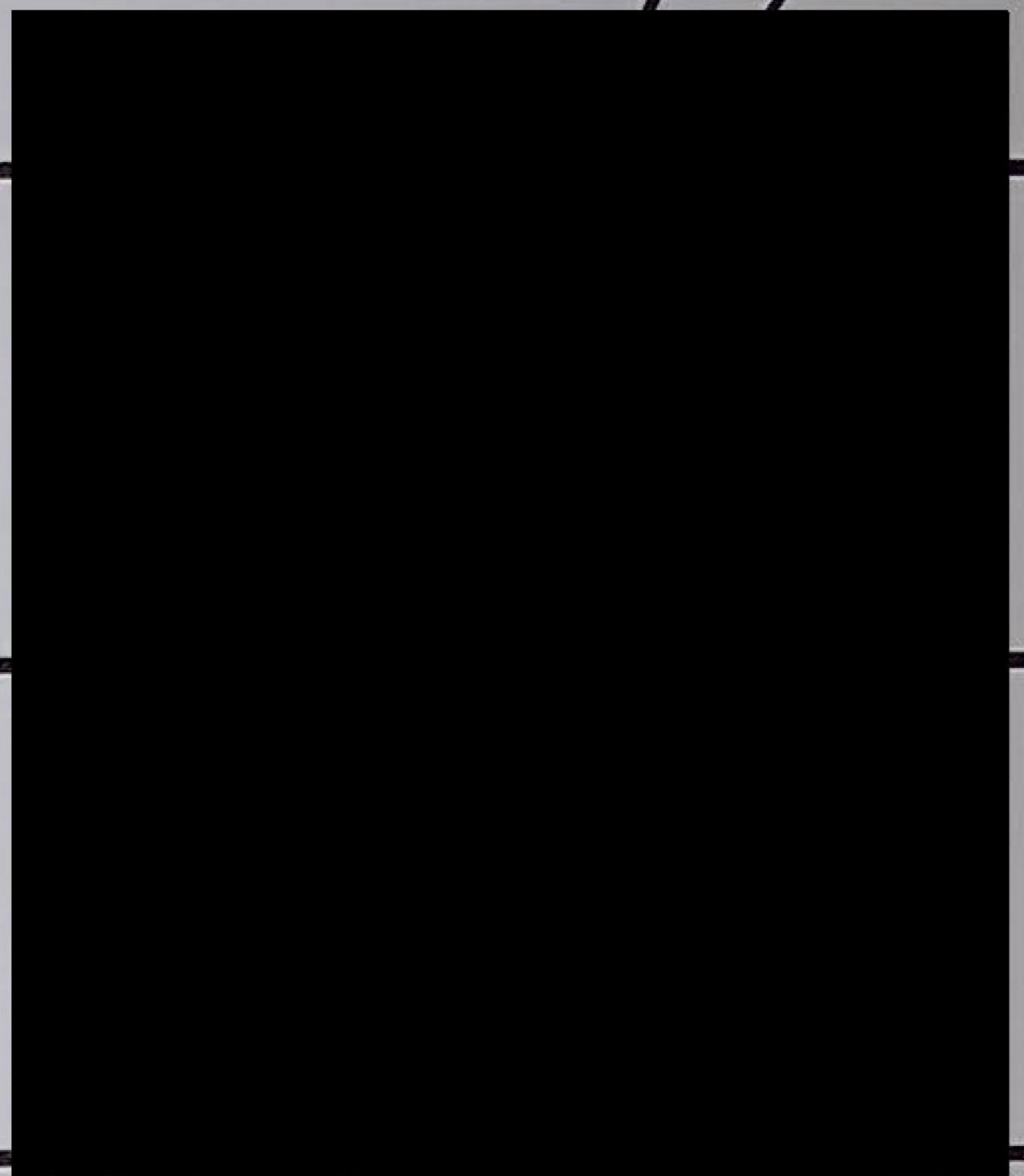
Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
79-08-03 C 06/06/1979	TO PREVENT ELECTRICAL SYSTEM FAILURE, SMOKE IN THE COCKPIT, AND/OR FIRE IN THE WIRE BUNDLE BEHIND THE INSTRUMENT PANEL	-- Hrs: UNKN C: --	PCW	No	D: -- Hrs: -- C: --	/
79-10-14 R1 05/30/1988	TO PROVIDE AN ALTERNATE SOURCE OF FUEL TANK VENTING IN CASE OF FUEL TANK VENT OBSTRUCTION BY FOREIGN MATERIAL, CONTD.	-- Hrs: -- C: --	N/A BY S/N	No	D: -- Hrs: -- C: --	/
79-25-07 12/13/1979	TO PRECLUDE THE POSSIBILITY OF ELECTRICAL OR ELECTRONIC COMPONENT DAMAGE OR AN IN-FLIGHT FIRE, CONTD.	05/24/1980 Hrs: 1252.9 C: --	PCW	No	D: -- Hrs: -- C: --	/
83-13-01 08/01/1983	TO ALERT THE PILOT TO THE POTENTIAL EFFECTS OF IMPROPER FUEL CAP SEALING	-- Hrs: -- C: --	N/A RAISED CAPS INSTALLED	No	D: -- Hrs: -- C: --	/
83-17-06 09/01/1983	TO PREVENT POSSIBLE DESTRUCTIVE AILERON FLUTTER	-- Hrs: -- C: --	N/A NOT INSTALLED	No	D: -- Hrs: -- C: --	/
84-10-01 R1 07/05/1988	TO PREVENT POWER LOSS OR ENGINE STOPPAGE DUE TO WATER CONTAMINATION OF FUEL SYSTEM	-- Hrs: -- C: --	N/A FUEL CAPS NOT INSTALLED	No	D: -- Hrs: -- C: --	/
86-24-07 01/07/1987	TO PREVENT ENGINE POWER INTERRUPTION DUE TO LOSS OF ATTACHMENT OF THE ENGINE CONTROLS	05/02/1987 Hrs: 345.6 C: --	PCW	No	D: -- Hrs: -- C: --	/
87-20-03 R2 09/24/1990	Superseded by 2011-10-09	-- Hrs: -- C: --	-----	No	D: -- Hrs: -- C: --	/
96-12-22 07/31/1996	TO PREVENT LOSS OF ENGINE OIL CAUSED BY LOOSE OR SEPARATED OIL FILTER ADAPTERS, WHICH COULD RESULT IN ENGINE, CONTD.	11/18/2014 Hrs: -- C: --	N/A NOT INSTALLED	No	D: -- Hrs: -- C: --	/

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
97-01-13 02/03/1997	TO PREVENT FUEL, OIL, OR HYDRAULIC SYSTEMS FAILURE CAUSED BY A COLLAPSED HOSE	-- Hrs: -- C: --	N/A NOT INSTALLED	No	D: -- Hrs: -- C: --	/
98-16-04 09/21/1998	TO PREVENT WING FAILURE DURING FLIGHT CAUSED BY THE ABSENCE OF AN ANGLE STIFFENER, WHICH COULD CAUSE LOSS OF,CONTD.	-- Hrs: -- C: --	N/A NOT INSTALLED	No	D: -- Hrs: -- C: --	/
2000-06-01 05/05/2000	To prevent foreign material from entering the fuel system and engine, which could result in loss of engine power,contd.	-- Hrs: -- C: --	N/A NOT INSTALLED	No	D: -- Hrs: -- C: --	/
2007-09-01 C2 04/25/2007	To detect and correct interference between the ground power electrical cable, the fuel strainer cable,contd.	-- Hrs: -- C: --	N/A NOT INSTALLED	No	D: -- Hrs: -- C: --	/
2008-02-18 02/28/2008	To prevent premature separation of the collar, which could result in the parachute failing to,contd.	-- Hrs: -- C: --	N/A NOT INSTALLED	No	D: -- Hrs: -- C: --	/
2008-10-02 05/12/2008	To prevent erroneous indications from the altimeter, airspeed, and vertical speed,contd.	-- Hrs: -- C: --	N/A NOT INSTALLED	No	D: -- Hrs: -- C: --	/
2008-26-10 C 01/05/2009	To prevent erroneous indications from the altimeter, airspeed, and vertical speed,contd.	-- Hrs: -- C: --	N/A NOT INSTALLED	No	D: -- Hrs: -- C: --	/
2011-10-09 06/17/2011	To prevent seat slippage or the seat roller housing from departing the seat rail, which may consequently cause,contd.	01/06/2021 Hrs: 3816.0 C: --	C/W BY INSPECTION AND MEASURE, DUE AT 3842.6	Yes	D: -- Hrs: 3916.0 C: --	/
2020-18-01 11/12/2020	[Recurring] To Detect and Address Cracking of the Wing Strut Attach Point. The Unsafe Condition, if Not Addressed, Could Result in Failure of the Wing in Operation, Which Could Result in Loss of Control of the Airplane	01/06/2021 Hrs: 3816 C: --	Due every 36 months or 1000 hrs. whichever comes first	Yes	D: -- Hrs: 4816 C: --	/

Category: Propeller

Position:

ATP Revision: 01/04/2021

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
82-27-02 R1 02/22/1983	TO PREVENT POSSIBLE PROPELLER BLADE SHANK FAILURE	-- Hrs: -- C: --	N/A BY BLADE S/N	No	D: -- Hrs: -- C: --	
2003-13-17 07/18/2003	To detect unsafe conditions that could result in separation of a propeller blade & loss of control,contd.	-- Hrs: -- C: --	N/A NOT INSTALLED	No	D: -- Hrs: -- C: --	
2005-14-11 08/17/2005	To prevent blade failure that could result in separation of a propeller blade and loss of control of the airplane	-- Hrs: -- C: --	N/A NOT INSTALLED	No	D: -- Hrs: -- C: --	
2006-24-07 01/03/2007	To detect potentially unsafe conditions that could result in a propeller blade separating from the hub,contd.	-- Hrs: -- C: --	N/A NOT INSTALLED	No	D: -- Hrs: -- C: --	

Category: Carburetors

Manufacturer: Marvel-Schebler

Model: MA-4-5

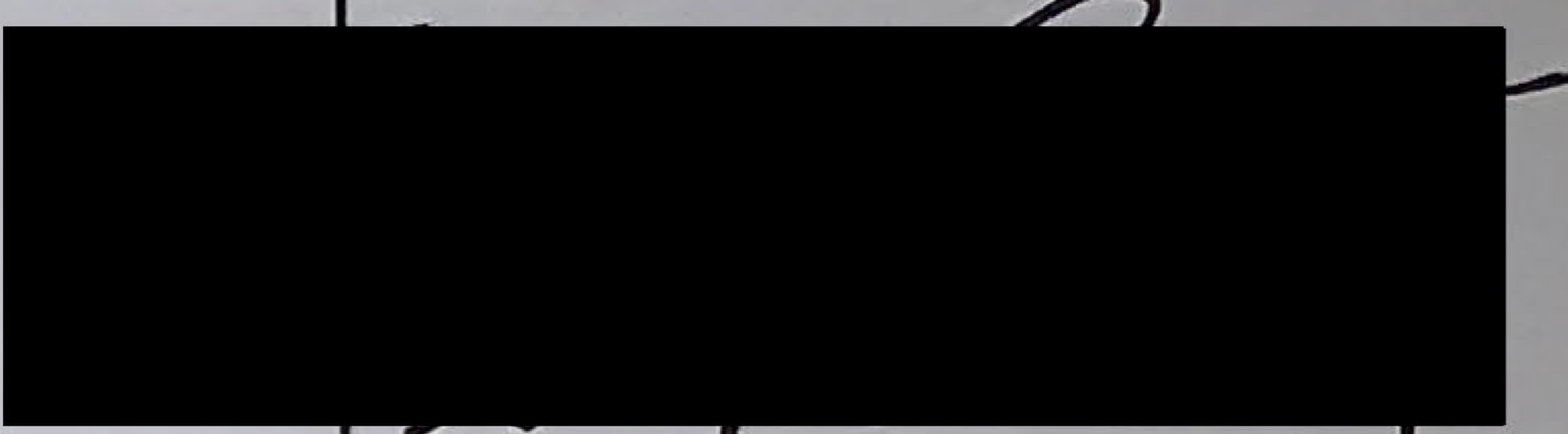
Position:

P/N: 10-4893-1

S/N: MS858010

ATP Revision: 12/13/2019

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
63-22-03 12/02/1963	ERRATIC ENGINE OPERATION OR COMPLETE ENGINE STOPPAGE	-- Hrs: -- C: --	N/A BY DATE	No	D: -- Hrs: -- C: --	
69-24-03 11/29/1969	TO PREVENT POSSIBLE POWER LOSSES DUE TO BLOCKAGE OF CARBURETOR METERING PASSAGES BY THREAD LUBRICANT, CONTD.	-- Hrs: -- C: --	N/A BY DATE	No	D: -- Hrs: -- C: --	
72-06-05 R2 07/03/1986	TO PREVENT LOOSENESS OR SEPARATION OF THE THROTTLE ARM	-- Hrs: -- C: --	N/A BY DATE	No	D: -- Hrs: -- C: --	

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
89-04-02 03/08/1989	TO PREVENT POSSIBLE JAMMING OF THE CARBURETOR THROTTLE	-- Hrs: -- C: --	N/A BY DATE	No	D: -- Hrs: -- C: --	

Category: Magnetos

Manufacturer: Bendix Corporation

Model: S-1200 SERIES

Position:

P/N: BL-349350-4 L/M

S/N: F13LA112R

ATP Revision: 12/12/2018

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
73-07-04 10/11/1973	Superseded by 94-01-03	-- Hrs: -- C: --	-----	No	D: -- Hrs: -- C: --	/
74-26-09 12/24/1974	S-20,-200,-1200 SERIES MAGNETOS	-- Hrs: -- C: --	N/A BY DATE	No	D: -- Hrs: -- C: --	/
78-09-07 R3 01/17/1983	Superseded by 96-12-07	-- Hrs: -- C: --	-----	No	D: -- Hrs: -- C: --	/
82-13-01 06/28/1982	MAGNETO AND ENGINE MALFUNCTION	-- Hrs: -- C: --	N/A BY DATE	No	D: -- Hrs: -- C: --	/
82-20-01 06/14/1983	TO PREVENT FAILURE OF IMPULSE COUPLING DUE TO IMPROPERLY HEAT TREATED (SOFT) FLYWEIGHTS RESULTING IN ENGINE, CONTD.	11/18/2014 Hrs: -- C: --	N/A BY DATE	No	D: -- Hrs: -- C: --	/
94-01-03 R2 06/28/1995	TO PREVENT MAGNETO FAILURE AND SUBSEQUENT ENGINE FAILURE	11/18/2014 Hrs: -- C: --	N/A BY DATE	No	D: -- Hrs: -- C: --	/

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
94-06-09 05/20/1994	TO PREVENT INJURY OR DEATH TO GROUND PERSONNEL DUE TO A NON- GROUNDED MAGNETO	11/18/2014 Hrs: -- C: --	N/A BY DATE	No	D: -- Hrs: -- C: --	
96-12-07 07/18/1996	Superseded by 2005-12-06	-- Hrs: -- C: --	-----	No	D: -- Hrs: -- C: --	
2005-12-06 07/19/2005	To prevent failure of the magneto impulse coupling assembly and possible engine failure	-- Hrs: -- C: --	N/A BY E/M	No	D: -- Hrs: -- C: --	

Category: Magnetos

Position:

ATP Revision: 12/12/2018

Manufacturer: Bendix Corporation

P/N: BL-349350-5 R/M

Model: S-1200 SERIES

S/N: F14AA085R

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
73-07-04 10/11/1973	Superseded by 94-01-03	-- Hrs: -- C: --	-----	No	D: -- Hrs: -- C: --	
74-26-09 12/24/1974	S-20,-200,-1200 SERIES MAGNETOS	-- Hrs: -- C: --	N/A BY DATE	No	D: -- Hrs: -- C: --	
78-09-07 R3 01/17/1983	Superseded by 96-12-07	-- Hrs: -- C: --	-----	No	D: -- Hrs: -- C: --	
82-13-01 06/28/1982	MAGNETO AND ENGINE MALFUNCTION	-- Hrs: -- C: --	N/A BY DATE	No	D: -- Hrs: -- C: --	

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
82-20-01 06/14/1983	TO PREVENT FAILURE OF IMPULSE COUPLING DUE TO IMPROPERLY HEAT TREATED (SOFT) FLYWEIGHTS RESULTING IN ENGINE, CONTD.	11/18/2014 Hrs: -- C: --	N/A BY DATE	No	D: -- Hrs: -- C: --	/
94-01-03 R2 06/28/1995	TO PREVENT MAGNETO FAILURE AND SUBSEQUENT ENGINE FAILURE	11/18/2014 Hrs: -- C: --	N/A BY DATE	No	D: -- Hrs: -- C: --	/
94-06-09 05/20/1994	TO PREVENT INJURY OR DEATH TO GROUND PERSONNEL DUE TO A NON-GROUNDED MAGNETO	11/18/2014 Hrs: -- C: --	N/A BY DATE	No	D: -- Hrs: -- C: --	/
96-12-07 07/18/1996	Superseded by 2005-12-06	-- Hrs: -- C: --	-----	No	D: -- Hrs: -- C: --	/
2005-12-06 07/19/2005	To prevent failure of the magneto impulse coupling assembly and possible engine failure	-- Hrs: -- C: --	N/A BY E/M	No	D: -- Hrs: -- C: --	/

Category: Engine

Manufacturer: Continental Motors

Model: O-470-R

Position:

P/N:

S/N: 1009683

ATP Revision: 12/12/2018

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
00-00-01 01/22/2001	Important for Cessna Oil Filter Adapter Assemblies listed in AD 96-12-22	-- Hrs: -- C: --	N/A NOT INSTALLED	No	D: -- Hrs: -- C: --	/
00-00-08 03/26/2002	Important notice for Teledyne Continental Motors reciprocating engines listed in AD 96-12-04	-- Hrs: -- C: --	N/A NOT INSTALLED	No	D: -- Hrs: -- C: --	/

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
56-06-01 07/01/1956	TO PREVENT SERIOUS ENGINE DAMAGE IN FLIGHT	-- Hrs: -- C: --	N/A BY DATE	No	D: -- Hrs: -- C: --	/
61-03-02 01/01/1961	Superseded by 63-15-01	-- Hrs: -- C: --	----- -----	No	D: -- Hrs: -- C: --	/
63-15-01 12/31/1969	EXHAUST VALVES	-- Hrs: -- C: --	N/A BY DATE	No	D: -- Hrs: -- C: --	/
73-13-04 06/20/1973	CARBURETOR BOWL DRAIN PLUG	-- Hrs: -- C: --	N/A BY DATE	No	D: -- Hrs: -- C: --	/
79-05-09 03/12/1979	TO PREVENT THE POSSIBLE LOSS OF OIL PRESSURE INDICATION	-- Hrs: -- C: --	N/A BY DATE	No	D: -- Hrs: -- C: --	/
85-08-02 04/15/1985	TO PREVENT POSSIBLE VALVE STEM WEAR, OIL CONTAMINATION, VALVE SEIZURE, AND TOTAL LOSS OF ENGINE POWER	-- Hrs: -- C: --	N/A BY DATE	No	D: -- Hrs: -- C: --	/
88-03-06 04/15/1988	TO PREVENT POSSIBLE LOSS OF ENGINE OIL AND SUBSEQUENT ENGINE FAILURE	-- Hrs: -- C: --	N/A BY DATE	No	D: -- Hrs: -- C: --	/
91-19-03 09/29/1991	TO PREVENT OPERATION WITH COLLAPSED OIL FILTER ELEMENTS, WHICH CAN RESULT IN LOSS OF OIL PRESSURE, CONTD.	-- Hrs: -- C: --	N/A BY DATE	No	D: -- Hrs: -- C: --	/
93-08-17 08/23/1993	TO PREVENT AN ENGINE FAILURE	-- Hrs: -- C: --	N/A BY DATE	No	D: -- Hrs: -- C: --	/

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
93-10-02 08/12/1993	TO PREVENT AN ENGINE FAILURE DUE TO A MISSING CYLINDER VALVE RETAINER KEY	-- Hrs: -- C: --	N/A BY DATE	No	D: -- Hrs: -- C: --	/
94-14-12 L 06/23/1994	Superseded by 95-21-15	-- Hrs: -- C: --	----- -----	No	D: -- Hrs: -- C: --	/
95-21-15 11/28/1995	TO PREVENT DETONATION DUE TO LOW OCTANE, WHICH CAN RESULT IN SEVERE ENGINE DAMAGE AND SUBSEQUENT FAILURE	-- Hrs: -- C: --	N/A BY DATE	No	D: -- Hrs: -- C: --	/
96-12-22 07/31/1996	[Recurring] TO PREVENT LOSS OF ENGINE OIL CAUSED BY LOOSE OR SEPARATED OIL FILTER ADAPTERS, WHICH COULD RESULT IN ENGINE, CONTD.	-- Hrs: -- C: --	N/A NOT INSTALLED	Yes	D: -- Hrs: -- C: --	/
97-15-01 L 07/30/1997	Superseded by 97-21-02	-- Hrs: -- C: --	----- -----	No	D: -- Hrs: -- C: --	/
97-21-02 10/27/1997	TO PREVENT EXTREME SIDE LOADING OF THE PISTON, AND CONSEQUENT FAILURE OF THE PISTON AND ENGINE	-- Hrs: -- C: --	N/A BY DATE	No	D: -- Hrs: -- C: --	/
98-01-08 E 12/23/1997	TO PREVENT EXHAUST ROLLER ROCKER ARM FAILURE, WHICH CAN RESULT IN BENT PUSH RODS, RUPTURED ROCKER ARM, CONTD.	-- Hrs: -- C: --	N/A NOT INSTALLED	No	D: -- Hrs: -- C: --	/
98-17-11 C 10/19/1998	TO PREVENT CRANKSHAFT FAILURE DUE TO CRACKING, WHICH COULD RESULT IN AN INFLIGHT ENGINE FAILURE AND POSSIBLE, CONTD.	-- Hrs: -- C: --	N/A NOT INSTALLED	No	D: -- Hrs: -- C: --	/
99-09-17 L 04/22/1999	Superseded by 99-19-01	-- Hrs: -- C: --	-----	No	D: -- Hrs: -- C: --	/

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
99-19-01 09/30/1999	To prevent crankshaft failure due to crankshaft cheek cracks, which could result in total engine power loss,contd.	-- Hrs: -- C: --	N/A BY DATE	No	D: -- Hrs: -- C: --	/
2000-08-51 E 04/28/2000	Superseded by 2000-23-21	-- Hrs: -- C: --	----- ----	No	D: -- Hrs: -- C: --	/
2000-23-21 12/12/2000	To prevent crankshaft connecting rod journal fracture, which could result in total engine power,contd.	-- Hrs: -- C: --	N/A BY DATE	No	D: -- Hrs: -- C: --	/
2007-04-19 R1 05/07/2007	To prevent cylinder separation that can lead to engine failure, possible engine compartment fire, and,contd.	-- Hrs: -- C: --	N/A NOT INSTALLED	No	D: -- Hrs: -- C: --	/
2009-19-07 C2 10/07/2009	[Recurring] To prevent loss of engine power due to cracks in the cylinder head, possible engine failure, and fire in,contd.	-- Hrs: -- C: --	N/A NOT INSTALLED	Yes	D: -- Hrs: -- C: --	/
2009-24-51 E 11/16/2009	Superseded by 2009-24-52	-- Hrs: -- C: --	----- ----	No	D: -- Hrs: -- C: --	/
2009-24-52 E 11/18/2009	Superseded by 2010-11-04	-- Hrs: -- C: --	----- ----	No	D: -- Hrs: -- C: --	/
2010-11-04 06/16/2010	To prevent excessive hydraulic lifter wear, which can result in loss of engine power & loss of control of the airplane	-- Hrs: -- C: --	N/A NOT INSTALLED	No	D: -- Hrs: -- C: --	/
2016-16-12 09/15/2016	To prevent failure of the cylinder assemblies, which could lead to failure of the engine, in-flight,contd.	-- Hrs: -- C: --	N/A NOT INSTALLED	No	D: -- Hrs: -- C: --	/

Category: Ignition Systems

Manufacturer: ACS Products Company

Position:

P/N:

ATP Revision: 12/12/2018

Model: IGNITION SWITCHES

S/N:

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
93-05-06 04/29/1993	[Recurring] TO PREVENT FAILURE OF IGNITION SWITCHES.	05/30/2018 Hrs: 3167.2 C: --	P/C/W	Yes	D: -- Hrs: 5167.2 C: --	