



EagleMed LLC
6601 W. Pueblo
Wichita Kansas 67209
USA

Phone:

Airworthiness Directives Compliance Record

Company: EagleMed LLC

Aircraft Registration No: N1551C

Category: Airframe

Position:

ATP Revision: 03/10/2016

Manufacturer: Beech Aircraft Corp.

S/N: LJ-1365

Model: C90A

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur?	Next Due	Facility Cert. No/Type Authorized By Signed
87-04-24 04/14/1987	[Recurring] TO PRECLUDE MALFUNCTION OF THE ELEVATOR TRIM CABLE SYSTEM	09/17/2014 H:6427.80	Exported From Profile and Compliance: Method Of Compliance: Verified N/A by A/C S/N.	N		7EMA575M Part 135 Darrin Carlson Kevin Stewart
88-04-07 04/18/1988	TO PREVENT POSSIBLE FAILURE OF THE WING LOWER FORWARD SPAR ATTACHMENT	09/17/2014 H:6427.80	Exported From Profile and Compliance: Method Of Compliance: Verified N/A by A/C S/N.	N		7EMA575M Part 135 Darrin Carlson Kevin Stewart
89-22-14 11/17/1989	TO PREVENT POSSIBLE AILERON FLUTTER	09/17/2014 H:6427.80	Exported From Profile and Compliance: Method Of Compliance: Verified N/A by A/C S/N.	N		7EMA575M Part 135 Darrin Carlson Kevin Stewart
89-25-10 01/04/1990	[Recurring] TO DETECT POSSIBLE FATIGUE CRACKING OF THE WING MAIN SPAR LOWER CAP AND ASSOCIATED STRUCTURE	09/17/2014 H:6427.80	Exported From Profile and Compliance: Method Of Compliance: Verified N/A by A/C S/N.	Y		7EMA575M Part 135 Darrin Carlson Kevin Stewart

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90-08-17 05/07/1990	TO PREVENT WATER ACCUMULATION IN THE AFT FUSELAGE WHICH CAN FREEZE AND RESTRICT CONTROL MOVEMENT	09/17/2014 H:6427.80	Exported From Profile and Compliance: Method Of Compliance: Verified N/A by A/C S/N.	N		7EMA575M Part 135 Darrin Carlson Kevin Stewart
92-15-01 08/21/1992	TO PREVENT UNDETECTED FAILURE OF ENGINE TRUSS-TO- FIREWALL BOLTS, WHICH COULD EVENTUALLY LEAD TO SEPARATION, CONTD.	09/17/2014 H:6427.80	Exported From Profile and Compliance: Method Of Compliance: Verified N/A by A/C S/N.	N		7EMA575M Part 135 Darrin Carlson Kevin Stewart
93-17-01 10/12/1993	TO PREVENT FAILURE OF THE LEADING EDGE FUEL CELL NIPPLES CAUSED BY CRACKING AND FUEL LEAKS, CONTD.	09/17/2014 H:6427.80	Exported From Profile and Compliance: Method Of Compliance: Verified N/A by A/C S/N.	N		7EMA575M Part 135 Darrin Carlson Kevin Stewart
96-11-12 06/24/1996	TO PREVENT POSSIBLE FAILURE OF THE OPTIONAL BEECH ELECTRIC TRIM SYSTEM OR THE COLLINS AUTOPILOT SYSTEM, WHICH,CONTD.	10/05/1995 H:479.60	Exported From Profile and Compliance: Method Of Compliance: Verified C/W SB2361 by Aerodynamics, Inc.	N		7EMA575M Part 135 Darrin Carlson Kevin Stewart
97-06-06 C 05/09/1997	TO PREVENT INADVERTENT MOVEMENT OF THE PILOT OR COPILOT CHAIR, WHICH COULD RESULT IN LOSS OF,CONTD.	09/17/2014 H:6427.80	Exported From Profile and Compliance: Method Of Compliance: Verified N/A by A/C S/N.	N		7EMA575M Part 135 Darrin Carlson Kevin Stewart
97-23-17 12/29/1997	TO PREVENT A LOSS OF VACUUM TO DEPRESSURIZE THE AIRPLANE CABIN, WHICH COULD RESULT IN PERSONAL INJURY TO THE,CONTD.	03/09/1999 H:1254.60	Exported From Profile and Compliance: Method Of Compliance: Verified C/W by MSB 2676 at Stevens Aviation, Inc.	N		7EMA575M Part 135 Darrin Carlson Kevin Stewart

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Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur?	Next Due	Facility Cert. No/Type Authorized By Signed
97-25-01 01/11/1998	TO PREVENT OUTFLOW/SAFETY VALVE CRACKING AND CONSEQUENT FAILURE, WHICH COULD RESULT IN RAPID DECOMPRESSION OF,CONTD.	06/01/1998 H:1119.60	Exported From Profile and Compliance: Method Of Compliance: Verified N/A by outflow valves S/N's by Raytheon Aircraft Services, Inc.	N		7EMA575M Part 135 Darrin Carlson Kevin Stewart
97-25-03 01/21/1998	TO PREVENT NOSE DOWN PITCH AND A DESCENT RATE LEADING TO AIRCRAFT DAMAGE AND INJURY TO PERSONNEL CAUSED BY,CONTD.	03/09/1999 H:1254.60	Exported From Profile and Compliance: Method Of Compliance: Verified C/W by Stevens Aviation	N		7EMA575M Part 135 Darrin Carlson Kevin Stewart
98-04-24 03/13/1998	TO MINIMIZE THE POTENTIAL HAZARDS ASSOCIATED WITH OPERATING THE AIRPLANE IN SEVERE ICING CONDITIONS,CONTD.	09/17/2014 H:6427.80	Exported From Profile and Compliance: Method Of Compliance: C/W by installing AD in AFM	N		7EMA575M Part 135 Darrin Carlson Kevin Stewart
99-09-15 05/18/1999	To prevent any components or wiring from interfering with the flight control mechanism caused by inadequate,contd.	09/17/2014 H:6427.80	Exported From Profile and Compliance: Method Of Compliance: Verified N/A by A/C S/N.	N		7EMA575M Part 135 Darrin Carlson Kevin Stewart
99-09-10 06/07/1999	To prevent the inability to extend the landing gear with the hand pump caused by debris entering the landing,contd.	06/01/1998 H:1119.60	Exported From Profile and Compliance: Method Of Compliance: Verified C/W SB 32- 3073R1 by Raytheon Aircraft Services, Inc.	N		7EMA575M Part 135 Darrin Carlson Kevin Stewart
99-10-07 06/08/1999	To detect and correct interference between the elevator control cable and equipment under the,contd.	03/22/2000 H:1445.20	Exported From Profile and Compliance: Method Of Compliance: Verified C/W by Stevens Aviation.	N		7EMA575M Part 135 Darrin Carlson Kevin Stewart

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99-18-15 09/27/1999	To detect any improperly machined landing gear emergency hand pumps, which, if not removed from,contd.	09/17/2014 H:6427.80	Exported From Profile and Compliance: Method Of Compliance: Verified N/A by A/C S/N.	N		7EMA575M Part 135 Darrin Carlson Kevin Stewart
2001-11-03 07/20/2001	To protect the blower motor circuit and reduce the possibility of the emission of smoke or a burning odor,contd.	03/22/2000 H:1445.20	Exported From Profile and Compliance: Method Of Compliance: Verified C/W MSB 34-3269 by Stevens Aviation	N		7EMA575M Part 135 Darrin Carlson Kevin Stewart
2002-01-10 02/22/2002	[Recurring] To detect and replace cracked main landing gear torque knees, which could result in failure of the main,contd.	10/31/2006 H:3349.30	Exported From Profile and Compliance: Method Of Compliance: Verified C/W SB 32-3134 by Heritage Flight. Terminated	N		7EMA575M Part 135 Darrin Carlson Kevin Stewart
2002-09-12 C 06/21/2002	To correct the installation of rivets in the left-hand and right-hand nacelle and spar assembly	12/18/2000 H:1512.20	Exported From Profile and Compliance: Method Of Compliance: Verified PCW SB 54-3308 by Stevens Aviation	N		7EMA575M Part 135 Darrin Carlson Kevin Stewart
2003-02-03 03/07/2003	To assure that clear and complete operating instructions are visible for opening the airstair,contd.	03/10/2003 H:1799.30	Exported From Profile and Compliance: Method Of Compliance: Verified PCW by SB 52-3096 by Stevens Aviation.	N		7EMA575M Part 135 Darrin Carlson Kevin Stewart
2003-13-16 08/25/2003	To detect and correct damage to the aft pressure bulkhead of the fuselage	09/17/2014 H:6427.80	Exported From Profile and Compliance: Method Of Compliance: Verified N/A by A/C S/N.	N		7EMA575M Part 135 Darrin Carlson Kevin Stewart
2004-17-02 10/04/2004	[Recurring] To detect and correct loose bolts not securing the pedestal cross shaft, which could result in limited,contd.	03/09/2004 H:1938.60	Exported From Profile and Compliance: Method Of Compliance: Verified C/W by MSB 27-3634 at Stevens Aviation	Y		7EMA575M Part 135 Darrin Carlson Kevin Stewart
2004-23-02 C 12/23/2004	[Recurring] To detect and correct cracks in the Nose Landing Gear (NLG) fork, which could result in reduced structural,contd.	09/17/2014 H:6427.80	Exported From Profile and Compliance: Method Of Compliance: Verified N/A by A/C S/N.	Y		7EMA575M Part 135 Darrin Carlson Kevin Stewart

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Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur?	Next Due	Facility Cert. No/Type Authorized By Signed
2005-01-04 C2 02/22/2005	To prevent fuel flow interruption, which could lead to uncommanded loss of engine power & loss of control,contd.	10/26/1998 H:1196.00	Exported From Profile and Compliance: Method Of Compliance: Verified C/W by MSB 2718R1 at Raytheon Aircraft Services, Inc.	N		7EMA575M Part 135 Darrin Carlson Kevin Stewart
2006-13-10 08/07/2006	To prevent water accumulation in the rudder trim tab, which could result in a change in the mass properties,contd.	09/18/2006 H:4226.40	Exported From Profile and Compliance: Method Of Compliance: Verified C/W by SB 55 -2365 by Heritage Flight	N		7EMA575M Part 135 Darrin Carlson Darrin Carlson
2006-13-10 08/07/2006	To prevent water accumulation in the rudder trim tab, which could result in a change in the mass properties,contd.	06/30/2006 H:3141.60	Exported From Profile and Compliance: Method Of Compliance: Verified C/W by SB 55 -2365 by Heritage Flight	N		7EMA575M Part 135 Darrin Carlson Kevin Stewart
2007-12-06 07/16/2007	To detect and correct improperly assembled or damaged flight controls, which could result in an,contd.	09/17/2014 H:6427.80	Exported From Profile and Compliance: Method Of Compliance: Verified N/A by A/C S/N.	N		7EMA575M Part 135 Darrin Carlson Kevin Stewart
2009-21-10 12/17/2009	AVOX SYSTEMS 3HT BOTTLE REPLACEMENT	04/01/2016 H:7586.30 C:8578.00	N/a by Oxygen Bottle Serial Number	N		Eaglemed LLC 7EMA575M Part 135 Darrin Carlson Darrin Carlson

Category: Engine

Position: 1

ATP Revision: 01/10/2017

Manufacturer: Pratt & Whitney Canada

S/N: PCE-25774

Model: PT6A-21

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur?	Next Due	Facility Cert. No/Type Authorized By Signed
80-08-12 04/18/1980	[Recurring] TO PRECLUDE WEAR OF THE FUEL PUMP DRIVE COUPLING SPLINES WHICH RESULTS IN AN ENGINE FLAMEOUT CONDITION	05/24/2007 H:3551.50	Exported From Profile and Compliance: Method Of Compliance: Verified Found embodied at Overhaul	N		7EMA575M Part 135 Darrin Carlson Kevin Stewart

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Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur?	Next Due	Facility Cert. No/Type Authorized By Signed
92-15-11 08/17/1992	TO PREVENT EXCESSIVE ENGINE ACCELERATION TIME THAT COULD RESULT IN AN AIRCRAFT'S INABILITY TO SAFELY, CONTD.	05/24/2007 H:3551.50	Exported From Profile and Compliance: Method Of Compliance: Verified N/A at Overhaul	N		7EMA575M Part 135 Darrin Carlson Kevin Stewart
97-04-12 03/14/1997	TO PREVENT ENGINE POWER REDUCTION DUE TO MALFUNCTION OF THE COMPRESSOR BOV, WHICH COULD RESULT IN A FORCED,CONTD	05/24/2007 H:3551.50	Exported From Profile and Compliance: Method Of Compliance: Verified embodied at Overhaul	N		7EMA575M Part 135 Darrin Carlson Kevin Stewart
2002-23-13 12/31/2002	[Recurring] To prevent failure of the turbine exhaust duct due to cracking that could result in possible separation of,contd.	05/24/2007 H:3551.50	Exported From Profile and Compliance: Method Of Compliance: Verified C/W at Overhaul	Y		7EMA575M Part 135 Darrin Carlson Kevin Stewart
2014-11-05 08/05/2014	To prevent uncontained engine failure and damage to the airplane	09/17/2014 H:6413.40	Exported From Profile and Compliance: Method Of Compliance: Verified N/A by Engine S/N	N		7EMA575M Part 135 Darrin Carlson Kevin Stewart

Category: Engine

Position: 2

ATP Revision: 01/10/2017

Manufacturer: Pratt & Whitney Canada

S/N: PCE-25765

Model: PT6A-21

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80-08-12 04/18/1980	[Recurring] TO PRECLUDE WEAR OF THE FUEL PUMP DRIVE COUPLING SPLINES WHICH RESULTS IN AN ENGINE FLAMEOUT CONDITION	05/25/2007 H:3551.50	Exported From Profile and Compliance: Method Of Compliance: Verified embodied at Overhaul	Y		7EMA575M Part 135 Darrin Carlson Kevin Stewart

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Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur?	Next Due	Facility Cert. No/Type Authorized By Signed
92-15-11 08/17/1992	TO PREVENT EXCESSIVE ENGINE ACCELERATION TIME THAT COULD RESULT IN AN AIRCRAFT'S INABILITY TO SAFELY, CONTD.	05/25/2007 H:3551.50	Exported From Profile and Compliance: Method Of Compliance: Verified N/A at Overhaul	N		7EMA575M Part 135 Darrin Carlson Kevin Stewart
97-04-12 03/14/1997	TO PREVENT ENGINE POWER REDUCTION DUE TO MALFUNCTION OF THE COMPRESSOR BOV, WHICH COULD RESULT IN A FORCED,CONTD	05/25/2007 H:3551.50	Exported From Profile and Compliance: Method Of Compliance: Verified C/W at Overhaul BOV N/A	N		7EMA575M Part 135 Darrin Carlson Kevin Stewart
2002-23-13 12/31/2002	[Recurring] To prevent failure of the turbine exhaust duct due to cracking that could result in possible separation of,contd.	05/25/2007 H:3551.50	Exported From Profile and Compliance: Method Of Compliance: Verified C/W at Overhaul	Y		7EMA575M Part 135 Darrin Carlson Kevin Stewart
2014-11-05 08/05/2014	To prevent uncontained engine failure and damage to the airplane	09/17/2014 H:6413.40	Exported From Profile and Compliance: Method Of Compliance: Verified N/A by Engine S/N	N		7EMA575M Part 135 Darrin Carlson Kevin Stewart

Category: Propeller

Position: 1

ATP Revision: 03/10/2016

Manufacturer: Hartzell Propeller

S/N: FY-3166

Model: HC-D4N-3C

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur?	Next Due	Facility Cert. No/Type Authorized By Signed
96-15-04 07/29/1996	TO PREVENT PROPELLER BLADE SEPARATION CAUSED BY PROPELLER BLADE SHANK CRACKS EMANATING FROM FORGING FLAWS,CONTD.	03/20/2014 H:954.90	Exported From Profile and Compliance: Method Of Compliance: Verified C/W at Overhaul	N		7EMA575M Part 135 Darrin Carlson Kevin Stewart

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Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur?	Next Due	Facility Cert. No/Type Authorized By Signed
2005-14-11 08/17/2005	To prevent blade failure that could result in separation of a propeller blade and loss of control of the airplane	03/20/2014 H:954.90	Exported From Profile and Compliance: Method Of Compliance: Verified C/W at Overhaul	N		7EMA575M Part 135 Darrin Carlson Kevin Stewart

Category: Propeller

Position: 2

ATP Revision: 03/10/2016

Manufacturer: Hartzell Propeller

S/N: FY-3198

Model: HC-D4N-3C

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur?	Next Due	Facility Cert. No/Type Authorized By Signed
96-15-04 07/29/1996	TO PREVENT PROPELLER BLADE SEPARATION CAUSED BY PROPELLER BLADE SHANK CRACKS EMANATING FROM FORGING FLAWS,CONTD.	04/02/2014 H:5264.90	Exported From Profile and Compliance: Method Of Compliance: Verified AD's researched at Overhaul	N		7EMA575M Part 135 Darrin Carlson Kevin Stewart
2005-14-11 08/17/2005	To prevent blade failure that could result in separation of a propeller blade and loss of control of the airplane	04/02/2014 H:5264.90	Exported From Profile and Compliance: Method Of Compliance: Verified AD's researched at Overhaul	N		7EMA575M Part 135 Darrin Carlson Kevin Stewart

Report Filters:

Complied Duration: All

Generated By: Darrin Carlson

Aircraft Registration No: N1551C