

Inspected all doors, panels, and structures for damage and security, cleaned and lubed all wheel bearings, lubed all control surfaces, checked all external lighting, checked seats, belts, and cockpit equipment, insp mlg tires additional work performed:

Performed four cycles of landing gear retraction. No leaks noted.

Serviced emergency gear extension nitrogen bottle.

Serviced Brake and Hydraulic reservoirs.

Installed repaired left engine fuel shut-off valve

Operational check performed on all flight controls, and spoiler system.

Operational check performed on electrical system, battery serviced, alternators check normal.

Installed serviceable dual vacuum indicator.

TTAF 7521,7

Date 22 SEP 2015 ACFT A12655B TACH 0.1 HOBBS

I certify this AIRCRAFT airworthy

IAW a ANNUAL inspection and
is approved for return to service

Aircraft approved for return to service for a maintenance test/training flight.

Items recommended for check:

1. Pressurization, Max and Min differential, and pressure relief valve operation.
2. Engine cooling, oil temperature, oil pressure within flight limits.
3. Engine manifold pressure/fuel flow IAW performance tables, throttle/prop/mixt alignment
4. Fuel system, Left and Right engine feed in all configurations.
5. Electrical charging system under typical flight load.
6. Stall characteristics with spoilers deployed, in landing configuration.
7. Aircraft rigging.

Aircraft checks satisfactory.

[Signature] Pilot In Command

10-10-16 Date

4.0 Hobbs

N 2655B

I certify that the Altimeter and Static System tests, as required by FAR 91.411, Paragraphs A, B, and C, have been performed.

The altimeter P/N 5934PA-3 S/N 447883

was tested to 30,000 feet on 9/15/15

The ATC Transponder Test and Inspection, as required by FAR 91.413, IAW FAR 43, Appendix F, have been performed.

Model KT76A S/N 130735

Signed [Redacted] Date 9/15/15

N 2655B

#2 TRANSPONDER

The ATC Transponder Test and Inspection, as required by FAR91.413, IAW FAR 43, Appendix F, have been performed.

Model KT76A S/N 115671

Signed [Redacted] Date 9/15/15

For Waukegan Avionics CRS# [Redacted]
Waukegan, Illinois [Redacted]



- Removed, repaired, and reinstalled the ARC CA-550A/FD Autopilot Computer, S/N 7403. Work performed referencing the Cessna Service Manual No: 7010434-3, Chg 3.
- Repaired the GNS 430W wiring for correct Cessna 800 IFCS interface referencing the GNS 430W Installation Manual, P/N 190-00356-08, Rev L.
- Removed the defective Edo-Aire NSD 360 HSI, P/N 52D137-2331, S/N 656B and replaced with an overhauled unit, S/N 1-7137B.
 - o Performed Autopilot system ground operation checks referencing the Sperry Systems Manual No. 7010434-2, Chg 1 – No discrepancies noted.
- Repaired the KMD 150 / GNS 430W interface wiring referencing the GNS 430W Installation Manual, P/N 190-00356-08, Rev L and the KMD 150 Installation Manual, P/N 006-10607-0000 Rev 0.

All items were ground checked in accordance with their respective manuals for proper operation and to ensure no adverse reactions occurred to other onboard systems or equipment. The above work was performed in accordance with current FAA regulations and in respect to work performed on the [REDACTED] aircraft.

FOR FAA CRS YXXR387Y



- Removed the bad left & right hand NAV blade antennas and replaced with new Comant CI 120-1 blades, S/N 506149 (LH) and 506150 (RH). Installed the CI120-3 Combiner, S/N 503930 in the previously used Cessna location and installed new coax leads. Performed ground operation checks – no discrepancies noted.
 - o Work performed per the Comant Install Instructions, ECH 08-102, Rev E and AC 43.13-2B, Chapter 3.
- Aligned the ARC CA-550A/FD Autopilot Computer, S/N 7403 to the G-1050A Attitude Gyro, S/N 1362G referencing the alignment procedures in the Cessna Service/Parts Manual No.: 7010434-3, Chg 3.

All items were ground checked in accordance with their respective manuals for proper operation and to ensure no adverse reactions occurred to other onboard systems or equipment. The above work was performed in accordance with current FAA regulations and in respect to work performed on the [REDACTED] aircraft.

FOR FAA CRS YXXR387Y



Altimeter Scale Correction Card *CO-PILOT*

Altimeter P/N: 5934PA-1A.86

Serial Number: 9D976

REFERENCE ALTITUDE IN FT	ALTIMETER READS	REFERENCE ALTITUDE IN FT	ALTIMETER READS
-1,000	-1,020	18,000	17,970
0	-15	20,000	19,960
500	485	22,000	21,970
1,000	990	25,000	24,960
1,500	1,495	30,000	29,875
2,000	1,990	35,000	34,970
3,000	2,990		
4,000	3,990		
5,000	4,980		
6,000	5,965		
8,000	7,985		
10,000	10,005		
12,000	12,010		
14,000	13,995		
15,000	14,990		
16,000	15,975		

Tested By: 

Date: 4/30/2015

MAKE: Cessna
MODEL: 421c
S/N: 421C0698
REG. NO: N2655B
WORK ORDER: 5497-06-2016



J.A. Air Center

DATE: 29 June, 2016
A/C TSN
HOBBS: 18.4

Airframe Entries

1 Aircraft weighed on certified jack point load cells, with full fuel, oil and hydraulic. 7 seats plus potty seat, all cabinets and dividers installed. All tear removed.

Aircraft as weighed,	Weight	Arm	Moment
LH Jack Point	1960.00	186.15	364854.00
RH Jack Point	2518.00	186.15	468725.70
Nose Jack Point	2230.00	99.52	221929.60
Usable fuel (206.0 gal)	-1236.00	165.20	-204187.20
Total as weighed	5472.00	155.58	851322.10

Max Take Off Wight 7450.00
New EWCG 155.58
New Empty Weight 5472.00
New Useful Load 1978.00

Maintenance Release

The aircraft, engine, appliance or component identified was repaired and inspected in accordance with current regulations of the Federal Aviation Administration and was found Airworthy for return to service. Pertinent details of the repair are as follows: 5497-06-2016, Dated 29 June, 2016.

DATE: 29 June, 2016

Work Order: 5497-06-2016

Printed by EBis 3 (datcomedia.com)

LEFT ENGINE LOGBOOK ENTRY

N2655B CESSNA 421C TCM GTSIO-520 HOBB:0000.2 WO#9857
12/4/15

Clean, inspect, run-up engine to detect fuel/oil leaks. Tightened and secured fuel vapor return line fitting. Tightened oil pressure line fitting at crankcase. Replaced turbo case oil drain line gasket P/N 649979 and turbo waste gate gasket P/N 652456. Run-up and leak check satisfactory.

With respect to the work performed this engine is approved for return to service.

Thomas Patt

Waukegan Aviation Services

AIRCRAFT LOGBOOK ENTRY

N2655B Make: CESSNA 421C S/N: 421C0698 HOBB: 0000.2 W/O# 9857
DATE: 12/4/15

De-fueled left wing tank, opened access panels, removed loose sealant inboard, cleaned inboard tank area and resealed with PR1422B-2. Replaced and resealed access panels. Re-fueled tank and leak checks satisfactory.

Serviced main gear struts with 5606 hydraulic fluid and dry nitrogen. Removed left brake caliper, cleaned, inspected and replaced (4) o-rings P/N MS28775-132. Replaced caliper, bled, leak and operational check good.

With respect to the work performed, this aircraft is approved for return to service.

for:

Waukegan Aviation Services, Inc.

AIRCRAFT LOGBOOK ENTRY

N2665B Make: CESSNA Model: C421C S/N: 421C0698 HM: WO: 9822

Date: 9/15/2015

Installed instrument panels previously removed. Replaced static system hoses to instruments. Installed previously wired GNS430W and indicator in racks. Repaired broken wire on KMD150. Installed post lights and repaired wiring as necessary. Repaired numerous static system leaks at fittings and selector valve. Static system checked and no leaks found. Replaced coax to #1 transponder antenna. Removed and replaced both engine analyzers after repairs. Installed KX165 after repairs at Bevan-Rabell (see FAA form 8130-3). Ran power and ground for LC-2 clock. Wired pilots side PTT switch. Repaired wiring at audio panel and checked operation. Replaced transistor for post light dimming system. Adjusted flap travel system to obtain correct full up position and intermediate position selections. Replaced day/night switch in dimmer selector and tested. Reconnected wires pulled from annunciator panel and replaced bad pins as necessary. Checked annunciator operation. Wired in post lights to throttle pedestal. Rewired spoilers to main bus from avionics bus. Repaired wires to ELT buzzer. Installed overhauled altimeter in Co-pilots side panel. (see FAA form 8130-3) from MidContinent Instruments. Installed battery in co-pilots clock. Replaced vacuum hoses behind instrument panel as necessary...end...

With respect to the work performed, this aircraft is approved for return to service.

for:
Waukegan Aviation Services, Inc

AIRCRAFT LOGBOOK ENTRY

N2655B Make: Cessna Model: 421C S/N: 421C0698 HM: WO# 9822

Date: 9/15/2015

Updated software in GNS430W, p/n 011-01060-40, s/n 97123944, per Garmin Software Service Bulletins No. 1461, Rev.A, to Main version 5.20, and to GPS version 5.0 per Garmin Software Service Bulletin No: 1317, Rev A. Checked configuration and revision pages in GNS 430W for correct versions.....end.....

With respect to the work performed, this aircraft is approved for return to service.

Howard W. Stedlecki
for:
Waukegan Avionics C.R.S # WOGR619K

WUKEGAN AVIONICS, INC.				N2655D			
WUKEGAN IL. 60087 CRS WOGR619K				Date: 9/15/2015			
TABLE 1 SCALE ERROR		TABLE 2 TEST TOLERANCES		TABLE 4 PRESSURE ALT. DIFF.		TABLE 3 FRICTION	
ALT	TOL	IND	TOL	IND	PRESS	HG	IND
-1000	20	-10	CASE LEAK	100 0	28.10	-1727	25 -7
500	20	15	HYSTERESIS		28.50	-1340	25 -5
1000	20	10	1st TEST POINT	75 21	29.00	-863	25 -3
1500	20	3	(50% MAX ALT)		29.50	-392	25 -2
2000	30	-3	2nd TEST POINT	75 18	29.92	0	25 0
3000	30	-5	(40% MAX ALT)		30.50	531	25 -1
4000	35	-3	AFTER EFFECT	30 -3	30.90	893	25 -3
6000	40	11			30.99	974	25 -4
8000	60	27					
10000	80	55					
12000	90	69	ALTIMETER TESTED TO	30000 Feet			
14000	100	85	DATE TESTED	9/15/2015			
16000	110	93	AIRCRAFT N NUMBER	N2655B			
18000	120	94	CUSTOMER	Sky King Flying			
20000	130	95	SERVICE ORDER #	14067/9822			
22000	140	103	ALTIMETER P/N	5934PA-3			
15000	155	97	ALTIMETER S/N	447883			
30000	180	85	ADC/ENCODER P/N	SD120			
35000	205		ADC/ENCODER S/N	55467			
40000	230		TRANSPONDER M				
45000	255		TRANSPONDER S				
50000	280		Tested in compliance				
53000	295		Authorized Signature				
55000	305						

M.B Aero

AMO 65-03

Registration: N2655B	Date: July 8, 2016
Serial #: 421C0698	Type: Cessna 421C
TTAF: 7726.0	TCAF:

Maintenance performed on unscheduled Cost Centre C03032-0001

Trouble-shot flap system, found broken roll pin inside of flap gearbox. Replaced roll pin IAW Standard Practices. Dual inspection carried out on flap system.

MB Aero Task Cards on Maintenance File

John Powell

July 8, 2016

The maintenance described above has been performed in accordance with the applicable standards of airworthiness.

Supreet Garcha

July 8, 2016

I certify that an independent inspection has been conducted on the controls identified.

AIRCRAFT LOGBOOK ENTRY

N2655B Make: Cessna Model: 421C S/N: 421C0698 HM: 31.5

Date: 8/4/2016
WO: 9947

Updated Software in NGT-9000, p/n 9029000-20000, s/n LXE00422, to Media p/n 9020010-004, R02.00, per L-3 Service Bulletin Number SB9029000-2, Rev A, Dated Nov 23, 2105 to be able to unlock the ATAS feature. Enabled the ATAS configuration and entered the unlock code C50C727D27 from L-3 Avionics Systems to unlock the feature. System check passed.

With respect to the work performed, this aircraft is approved for return to service.

Chadly Jensen

for:

Waukegan Avionics C.R.S #

AIRCRAFT LOGBOOK ENTRY

N2655B Make: CESSNA Model: 421C S/N: 421C0698 WO# 9922

DATE: 6/16/16 Removed and replaced flap motor with new, P/N C145-450-3. Lubed flap rollers, HOBB: 14.8 rod ends and actuator drive shafts. Operational checks satisfactory.

With respect to the work performed, this aircraft is approved for return to service.

Thomas Pall A&P

for:

Waukegan Aviation Services, Inc.

1. Approving Civil Aviation Authority/Country: FAA/United States		2. AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG			3. Form Tracking Number: 58247	
4. Organization Name and Address: MAYDAY AVIONICS, INC. 5500 44th st. S.E. Grand Rapids, MI. 49512					5. Work Order/Contract/Invoice Number: 58247	
6. Item:	7. Description:	8. Part Number:	9. Quantity:	10. Serial Number:	11. Status/Work:	
1	CA-550A/FD	42680-0004	1	7403	Aligned	
12. Remarks: Aligned to G-1050A gyro S/N 1362G referencing Cessna Service/Parts Manual# 7010434-3 ch. 3.						
13a. Certifies the items identified above were manufactured in conformity to: ☐ Approved design data and are in a condition for safe operation. ☐ Non-approved design data specified in Block 12.			14a. ☒ 14 CFR 43.9 Return to Service ☐ Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/Authorization No.:		14b. Authorized Signature:		14c. Approval/Certificate No.:
						
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yyyy):		14d. Name (Typed or Printed): Bob Weber		14e. Date (dd/mmm/yyyy): 19/Feb/2016
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article. Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1. Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States		2. AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG			3. Form Tracking Number: 58101	
4. Organization Name and Address: MAYDAY AVIONICS, INC. 5500 44th st. S.E. Grand Rapids, MI. 49512					5. Work Order/Contract/Invoice Number: 58101	
6. Item:	7. Description:	8. Part Number:	9. Quantity:	10. Serial Number:	11. Status/Work:	
1	CA-550A/FD	42680-0004	1	7403	Repaired	
12. Remarks: Repaired referencing Cessna service/Parts Manual# 7010434-3 ch. 3.						
13a. Certifies the items identified above were manufactured in conformity to: ☐ Approved design data and are in a condition for safe operation. ☐ Non-approved design data specified in Block 12.			14a. ☒ 14 CFR 43.9 Return to Service ☐ Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/Authorization No.:		14b. Approval/Signature:		14c. Approval/Certificate No.:
						
13d. Name (Typed or Printed):		13e. Date (dd/mm/yyyy):		14d. Name (Typed or Printed):		14e. Date (dd/mm/yyyy):
				Bob Weber		16/Feb/201
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article. Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1. Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/United States		2. AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG				3. Form Tracking Number: 000275657	
4. Organization Name and Address: 9400 E. 34th Street N. Wichita, KS 67226 FAA CRS OL2R061L						5. Work Order/Contract/Invoice Number 000275657	
6. Item:	7. Description:	8. Part number:	9. Quantity:	10. Serial number:	11. Status/Work:		
1	Navigation Situation Display	52D137-2331 IG-895A	1	1-7137B	Overhauled		
12. Remarks: THE WORK ORDER REFERENCED DESCRIBES THE ACTUAL WORK PERFORMED. ALL WORK WAS PERFORMED REF. MANUAL # 68S304 / 2 / 1 Dec 1996 Certifies that the work specified in Block 11/12 was carried out in accordance with EASA Part-145 and in respect to that work the component is considered ready for release to service under EASA Part-145 Approval Number: 145.4675							
13a. Certifies the items identified above were manufactured in conformity to: ☐ Approved design data and are in a condition for safe operation. ☐ Non-approved design data specified in Block 12.			14a. ☒ 14 CFR 43.9 Return to service ☒ Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, Part 43 and in respect to that work, the items are approved for return to service.				
13b. Authorized signature:		13c. Approval/Authorization No.:		14b. Authorized signature:		14c. Approval/Certificate No.:	
13d. Name (Types or Printed):		13e. Date (dd/mm/yyyy):		14d. Name (Types or Printed):		14e. Date (dd/mm/yyyy):	
				Randall K Ferguson		12/Feb/2016	
User/Installer Responsibilities							
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article. Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1. Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by user/installer before the aircraft may be flown.							

1. Approving Civil Aviation Authority/Country: FAA/United States		2. AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG			3. Form Tracking Number: 000256700	
4. Organization Name and Address:  MidContinent™ INSTRUMENTS & AVIONICS				5. Work Order/Contract/Invoice Number 000256700		
6. Item:		7. Description:		8. Part number:	9. Quantity:	10. Serial number:
1		ALTIMETER		5934PA-1A.86 5934PA-1	1	9D976
11. Status/Work: Overhauled						
12. Remarks: THE WORK ORDER REFERENCED DESCRIBES THE ACTUAL WORK PERFORMED. ALL WORK WAS PERFORMED REF. MANUAL # TM-5934 / REV. 4 / 1 Mar 2010						
Certifies that the work specified in Block 11/12 was carried out in accordance with EASA Part-145 and in respect to that work the component is considered ready for release to service under EASA Part-145 Approval Number: 145.4675						
13a. Certifies the items identified above were manufactured in conformity to: ☐ Approved design data and are in a condition for safe operation. ☐ Non-approved design data specified in Block 12.				14a. ☒ 14 CFR 43.9 Return to service ☒ Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, Part 43 and in respect to that work, the items are approved for return to service.		
13b. Authorized signature:		13c. Approval/Authorization No.:		14c. Approval/Certificate No.:		
13d. Name (Types or Printed):		13e. Date (dd/mm/yyyy):		14d. Name (Types or Printed):		14e. Date (dd/mm/yyyy):
				Nicholas Ryan Howe		12 May 2015
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.						
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.						
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority / Country: FAA / UNITED STATES		AUTHORIZED RELEASE CERTIFICATE FAA FORM 8130-3, AIRWORTHINESS APPROVAL TAG			3. Form Tracking Number: 59049	
4. Organization Name and Address: BEVAN-RABELL, INC. 1880 S AIRPORT ROAD WICHITA, KS 67209 CRS # KG2R951K					5. Work Order/Contract/Invoice Number: 59049-1	
6. Item:	7. Description:	8. Part Number:	9. Qty.:	10. Serial Number:	11. Status/Work:	
1	VHF TRANSCEIVER KX165	069-1025-25	1	63214	TESTED	
12. REMARKS: REFERENCE SERVICE SHOP REPORT FOR WORK PERFORMED: THIS UNIT WAS REPAIRED AND TESTED IN ACCORDANCE WITH PROCEDURES CONTAINED IN: BENDIX/KING MM # 006-05179-0009 REV. 9, MAY, 1999.						
13a. Certifies the items identified above were manufactured in conformity to: ☐ Approved design data and are in a condition for safe operation ☐ Non-approved design data specified in Block 12			14a. ☒ 14 CFR 43.9 Return to Service ☐ Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval/Authorization No.:		14b. Authorized Signature:		14c. Approval/Certificate No.:
13d. Name (Typed or Printed):		13e. Date (dd/mm/yyyy):		14d. Name (Typed or Printed):		14e. Date (dd/mm/yyyy):
				BILL C. BARNES		12/JUN/2015
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article. Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1. Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

1. Approving Civil Aviation Authority/Country: FAA/UNITED STATES		2. AUTHORIZED RELEASE CERTIFICATE FAA FORM 8130-3, AIRWORTHINESS APPROVAL TAG				3. Form Tracking Number: 186368	
4. Organization Name and Address: Aerospace Turbine Rotables Inc. 3414 W 29th St South Wichita, KS 67217 US Ph: (316) 943-6100, Fax: (316) 943-2917 FAA Approval Holder: NV2R045L						5. Work Order, Contract or Invoice Number: WO68679	
6. Item	7. Description:	8. Part Number:	9. Quantity:	10. Serial Number:	11. Status/Work:		
1	OXYGEN REGULATOR	801366-11	1.00	0703	OVERHAULED		
12. REMARKS Overhauled and certified IAW the documents listed. Full details of work performed held on work order. Manuals Used: Manual ID: 801366, Type: DWG, Rev: V, Rev Date: 12/12/2007; Manual ID: 7801366, Type: TP, Rev: D, Rev Date: 10/19/1993							
The work specified in Blocks 11/12 was performed in accordance with EASA-Part 145 and with respect to the work, the aircraft component is considered ready for release to service under EASA Approval Certificate EASA 145.4033.							
13a. Certifies the items identified above were manufactured in conformity to: ☐ Approved design data and are in a condition for safe operation. ☐ Non-approved design data specified in Block 12				14a. ☒ 14 CFR 43.9 Return to Service ☒ Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval Authorization No:		14c. Approval/Certificate No:			
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yy)		14d. Name (Typed or Printed): STEVEN L. STARK		14e. Date (dd/mmm/yyyy): 19/Mar/2015	
User/Installer Responsibilities							
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.							
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.							
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.							

1. Approving Civil Aviation Authority/Country: FAA/UNITED STATES		2. AUTHORIZED RELEASE CERTIFICATE FAA FORM 8130-3, AIRWORTHINESS APPROVAL TAG			3. Form Tracking Number: 186367	
4. Organization Name and Address: Aerospace Turbine Rotables Inc. 3414 W 29th St South Wichita, KS 67217 US Ph: (316) 943-6100, Fax: (316) 943-2917 FAA Approval Holder: NV2R045L				5. Work Order, Contract or Invoice Number: WO68678		
6. Item	7. Description:	8. Part Number:	9. Quantity:	10. Serial Number:	11. Status/Work:	
1	OXYGEN BOTTLE	11 CU FT	1.00	70737DG/0703	INSPECTED	
12. REMARKS Inspected and certified IAW the documents listed. Full details of work performed held on work order. Manuals Used:						
The work specified in Blocks 11/12 was performed in accordance with EASA-Part 145 and with respect to the work, the aircraft component is considered ready for release to service under EASA Approval Certificate EASA.145.4033.						
13a. Certifies the items identified above were manufactured in conformity to: ☐ Approved design data and are in a condition for safe operation. ☐ Non-approved design data specified in Block 12			14a. ☒ 14 CFR 43.9 Return to Service ☒ Other regulation specified in Block 12 Certifies that unless otherwise specified in Block 12, the work identified in Block 11 and described in Block 12 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
13b. Authorized Signature:		13c. Approval Authorization No:		14c. Approval/Certificate No:		
13d. Name (Typed or Printed):		13e. Date (dd/mmm/yy)		14d. Name (Typed or Printed): STEVEN L. STARK		
				14e. Date (dd/mmm/yyyy): 19/Mar/2015		
User/Installer Responsibilities						
It is important to understand that the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.						
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/article(s) from the airworthiness authority of the country specified in Block 1.						
Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.						

